

Design and Scaling Laws of a 40-MW class Electric-Wire-Interconnect-System (EWIS) for Liquid-H₂ Fuel-cell Propulsion

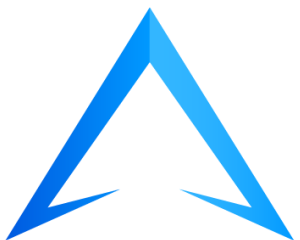
Dr. Timothy Haugan, *U.S. Air Force Research Laboratory*

Dr. Mary Ann Sebastian, Dr. Bang Hung Tsao, *University of Dayton Research Institute*

Dr. Chris Kovacs, *AFRL, Scintillating Solutions LLC*

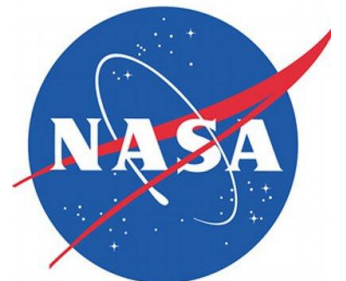
**2nd International Conference on Emissions Free Air Transport
Through Electric Propulsion**

Hybrid, 30,31 Aug 2022



CHEETA

Center for High-Efficiency Electrical
Technologies for Aircraft

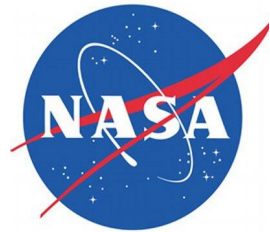


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given at EFATS 2022, August 30-31, 2022.



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Acknowledgment of Support

- NASA University Leadership Initiative (ULI)
#80NSSC19M0125
- AFRL/RQ Aerospace Systems Directorate
- AFRL/AFOSR LRIR #18RQCOR100, Dr. Ken Goretta



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Electric Aircraft Propulsion – the 3rd era in Aviation!



NASA X-57 Maxwell

www.nasa.gov/sites/default/files/thumbnails/image/x-57_maxwell_city.jpg

	Electric	2-12 Passenger	120-440 Passenger
Drivetrain Efficiency	90-95%	15-20%	35-40%
Energy Cost	\$2.55/gge @ \$0.07/kwh	AvGas \$6/gal	JP8, \$3.5-\$4/gal
Carbon-free Fuel	yes	costly	costly
Operation Costs	80-90% lower		
Maintenance		10-100x higher	10-100x higher
Altitude cruise	100-40,000 ft	10,000-20,000 ft	30,000--40,000 ft
Noise	70% lower		
Airports to Use	13,000+ small/regional		

Electric Aircraft Pre-Sales, as of 27 Apr 2022

Electric Aircraft Manufacturer	Model	Aircraft # seats	Vehicle Cost (each)	Goal for Operation (Year)	Company IPO, Total \$	Airline Purchases	# Sales	Total Sales (\$)	Reference Date	Reference
Archer Aviation			\$5M	2025	\$1.1B	United, Mesa	200 + 100 (option)	\$1B	10-Feb-2021	techcrunch.com/2021/02/10/
Beta	Alia-250		\$4.5M	2024	\$0.796B	United Parcel Service (UPS)	10 firm + 140 (option)	\$0.675B	20-Apr-2022	https://www.futureflight.aero/news-article/2022-04-20
Boom Aerospace	Overture 'supersonic'		\$200M	2029		United	15 + 35	\$3B	13-Dec-2021	www.prnewswire.com/news-releases
Bye Aerospace	eFlyer 2	2	\$0.49M			many	~ 380	\$0.19B	25-Apr-2022	https://byeaerospace.com/
Bye Aerospace	eFlyer 4	4	\$0.63M			"	~ 385	\$0.24B	25-Apr-2022	https://byeaerospace.com/
Bye Aerospace	eFlyer 800	8	\$6M			"	135	\$1.1B	25-Apr-2022	https://byeaerospace.com/
Eviation	Alice	6 to 9	\$4M			Cape Air, many	150	\$0.6B	18-Apr-2022	www.geekwire.com/2022/
Eviation	Alice	6 to 9	"	2026		DHL	12	\$0.05	18-Apr-2022	www.geekwire.com/2022/
Joby Aviation				2024	\$1.6B	Joby Aviation				
Lilium		7	\$4.5M		\$1.94B	Azul (Brazil), NetJets	370	\$1.7B	8-Mar-2022	https://robbreport.com/motors/aviation/
EVE Urban Air Mobility			\$8M			Azorra (Embraer), many	1,785	\$14.28	16-Mar-2022	/www.reuters.com/business/aerospace-defense
Heart Aerospace	ES-19	19 seats	\$8.8M	2026		United, Mesa	200	\$0.9B	7-Jul-2021	www.bloomberg.com/news/articles/2021-07-13/
Textron Aviation	Cessna Grand Caravan EX	10-14 seats	\$2.4M			Surf Air Mobility	150	\$0.36B	20-Jul-2021	media.txtav.com/201086
Velos Electro, Pipestrel	Velis Electro	2	\$0.094M			Europe, E-trainers	250	\$0.024B	8-Feb-2021	https://www.electrive.com/2021/02/08
Vertical Aerospace	VA-X4	5	\$4M	2024	\$2.2B	many	1,350	\$5.4B	4-Apr-2022	simpleflying.com/
Volocopter					\$0.376B				4-Mar-2022	https://www.electrive.com/2022/03/04
ZeroAvia	ZA2000-RJ engines, e.g. for UA CRJ-550	100	\$2M	2028		United	50 + 50 (engines)	\$0.1B	13-Dec-2021	www.prnewswire.com/news-releases
					\$8.00B	Subtotals	5,562 +	\$29.6B +		

Presales
> 5,550 aircraft*
> \$29B
IPO's > \$8B

- Compare to ~ 43,000 commercial aircraft worldwide in 2022
- Most Aircraft not built or certified yet

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Center for High Efficiency Electric Aircraft (CHEETA)

Liquid-H₂ Fuel Cells + Battery, Design Study, FY20-22

- **NASA University Leadership Initiative (ULI)**
- **U. of Illinois UC**, Dr. Phil Ansell, Lead PI
 - Boeing Research and Technology (BR&T)
 - General Electric Global Research (GE Global)
 - Massachusetts Institute of Technology (MIT)
 - University of Arkansas (UARK)
 - Rensselaer Polytechnic Institute (RPI)
 - Ohio State University (OSU)
 - University of Dayton Research Institute (UDRI)
 - AFRL/RQ - collaboration, no \$
- **Funding ~ \$6M for 3 years, FY20-22**



Boeing 737-800, CHEETA VO3 Comparison

178 Passengers

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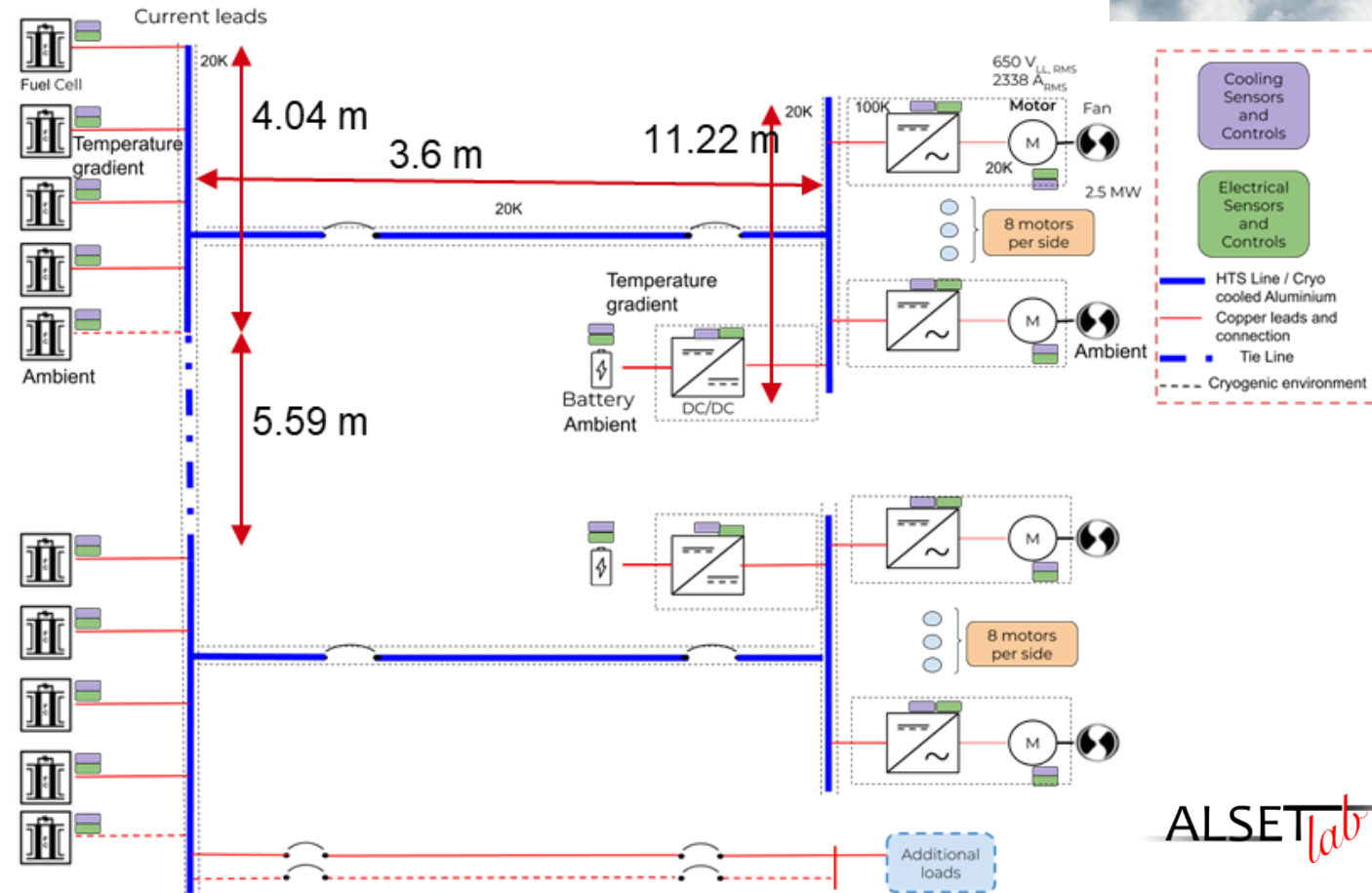
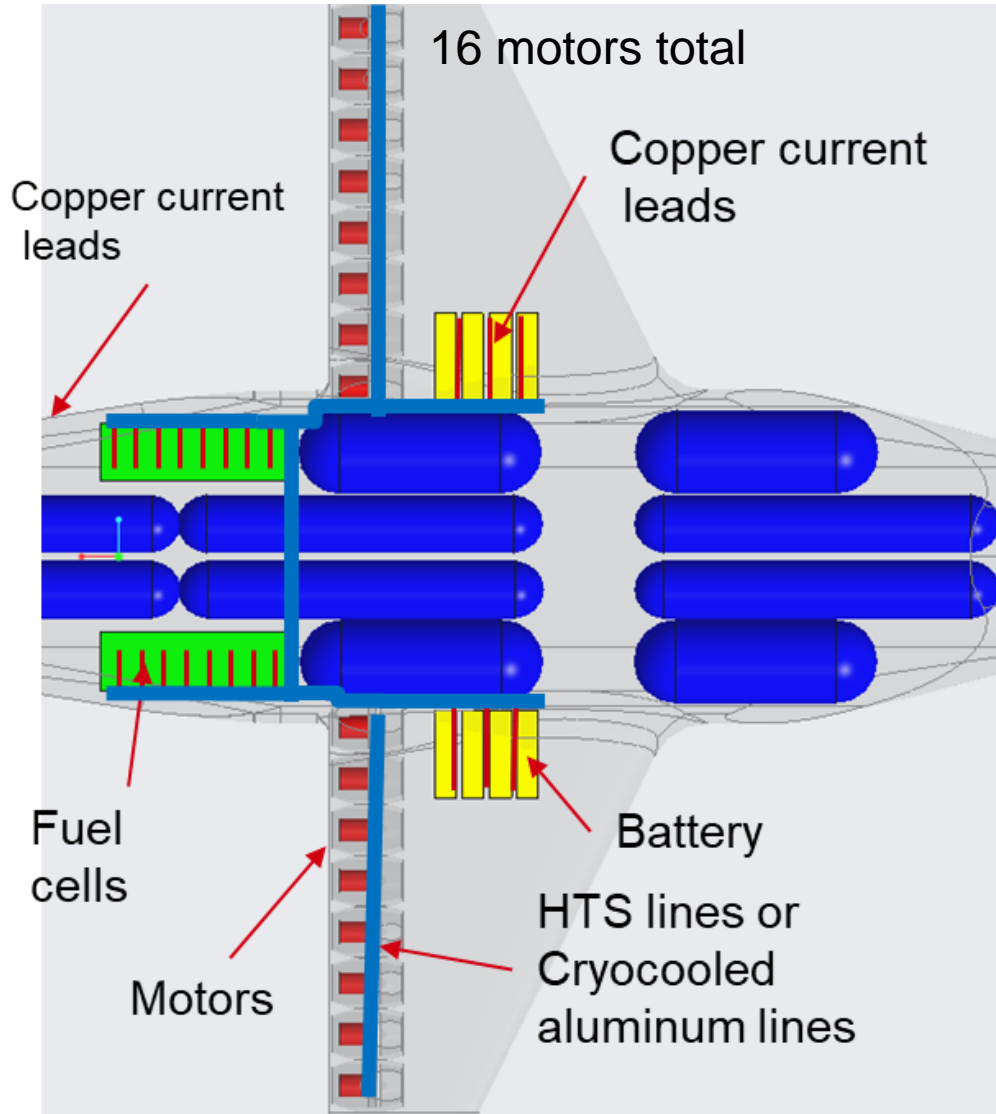
	737-800	CHEETA VO3
Propulsion Fuel, Energy	JP-8	Liquid-H ₂ , Li-Battery
Operational Empty Weight (OEW)	91,300 lb	138,000 lb
Maximum Take Off Weight (MTOW)	174,000 lb	190,000 lb
Fuel Weight	47,570 lb	16,700 lb
Fuel Energy	917 GJ	1,074 GJ
Peak Motor Power	24,000 – 27,000 lbf	22.5 MW
Cruise Motor Power	5,960 lbf	16.4 MW

https://en.wikipedia.org/wiki/Boeing_737_Next_Generation

CHEETA Liquid-H₂+fuel-cell, 2020 drivetrain architecture V02

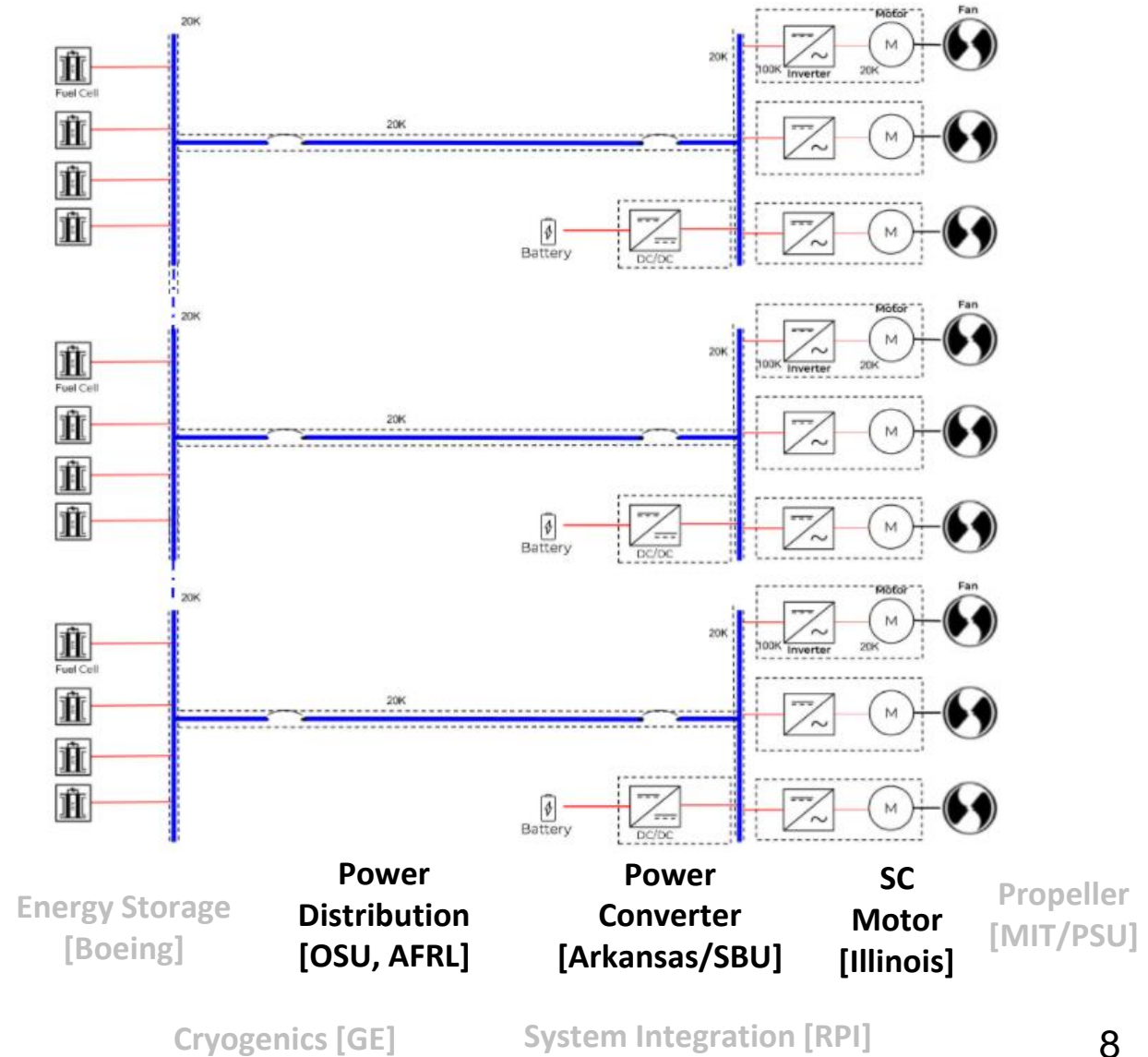
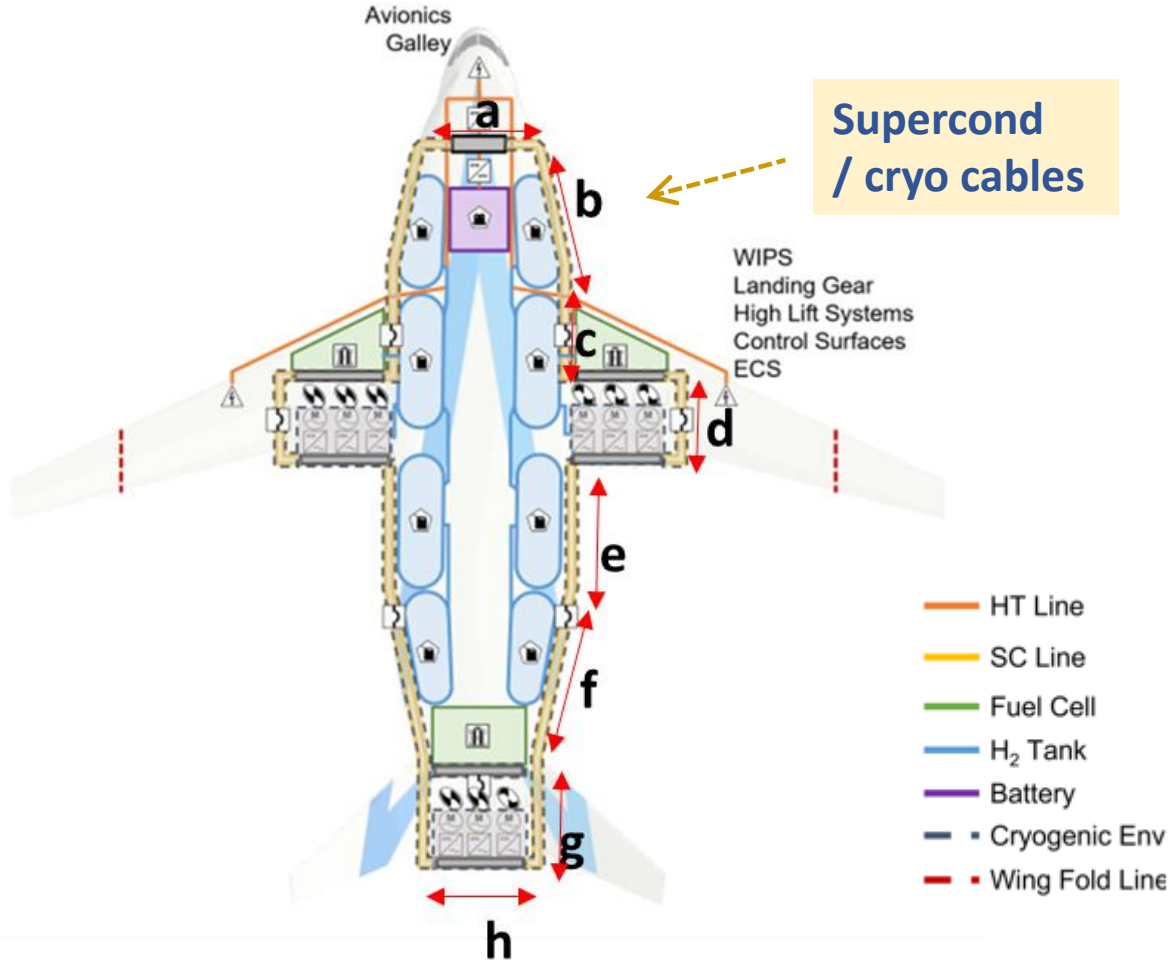
40 MW, 1 kV, 40 kA!, 43 m cable length

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CHEETA, Sept 2021 drivetrain architecture V03

22.5 MW, $\pm 270V$, (13.9 kA, $\sim 75.5m$ to/from for $\frac{1}{2}$ side) cable length



V03 tip-to-tip length = 41.91 m
 V03 length = 37.59 m
 Cable length one-side = $(b+c+d+e+f+g)+(a+h)/2 = 37.72$ m

AFRL/UDRI Project Task 1

Power Distribution System: 22.5 MW, \pm 270 V, 13.9 kA

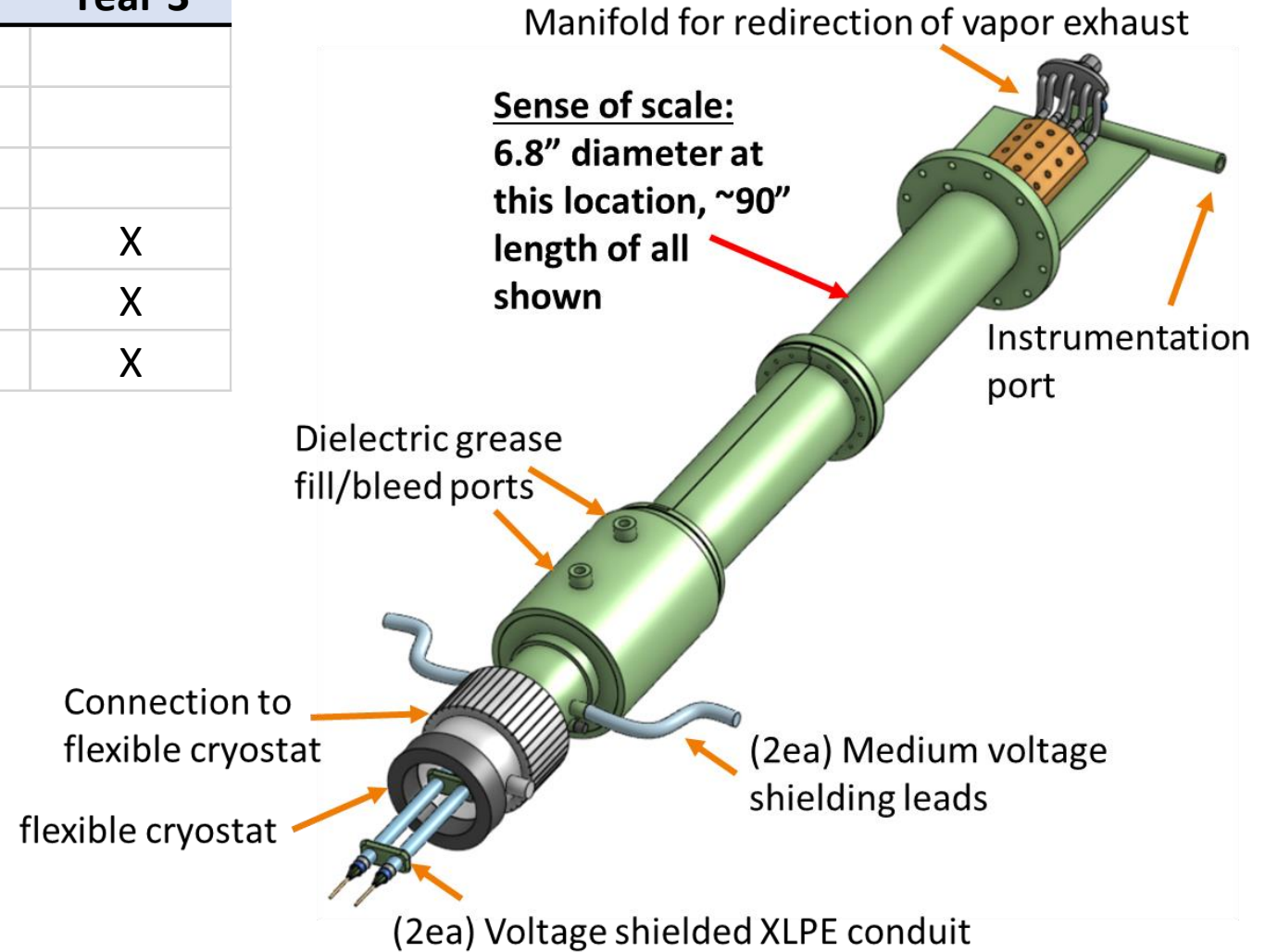
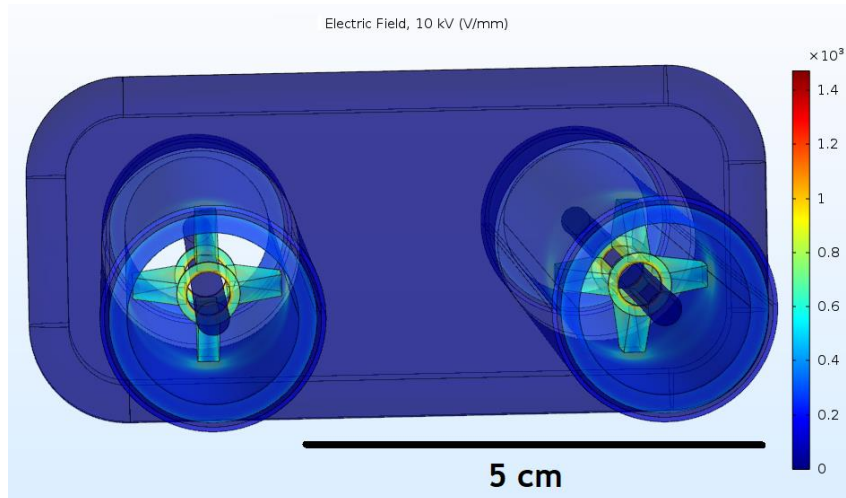
- **Determine Scaling Laws for Major Components:** how mass, heat-loads vary
- **Goals:** compare different technologies, design options, sensitivities, and swaps

				Year 1	Year 2	Year 3
Metal Cable	Al-99.999% 'hyperconductor'		20K, 300K	X		
	Copper-clad Aluminum (CCA)		"	X		
Metal Cable Conduit			300 K		X	X
Superconductor Cable	YBCO, MgB ₂		20K	X		
Current Leads	Metals, Cu or Al		300K <--> 77K	X		
Current Leads	Superconductor		77K <--> 20K	X		
Busbar	Superconductor or Metal		20K, 300K	X	X	
Cryoflex Tubing	Nexans, custom		20-80 K	X		X
T-Joint	YBCO <--> Copper-clad-Al		20-30 K		X	
High Voltage Cables			0-10 kV		X	X
Low --> High Voltage Insulation			0-10 kV			X
Circuit Breakers			DC			X
Switches, Fault Current Limiters						X
System Integration					X	X

AFRL/UDRI Project Task 2

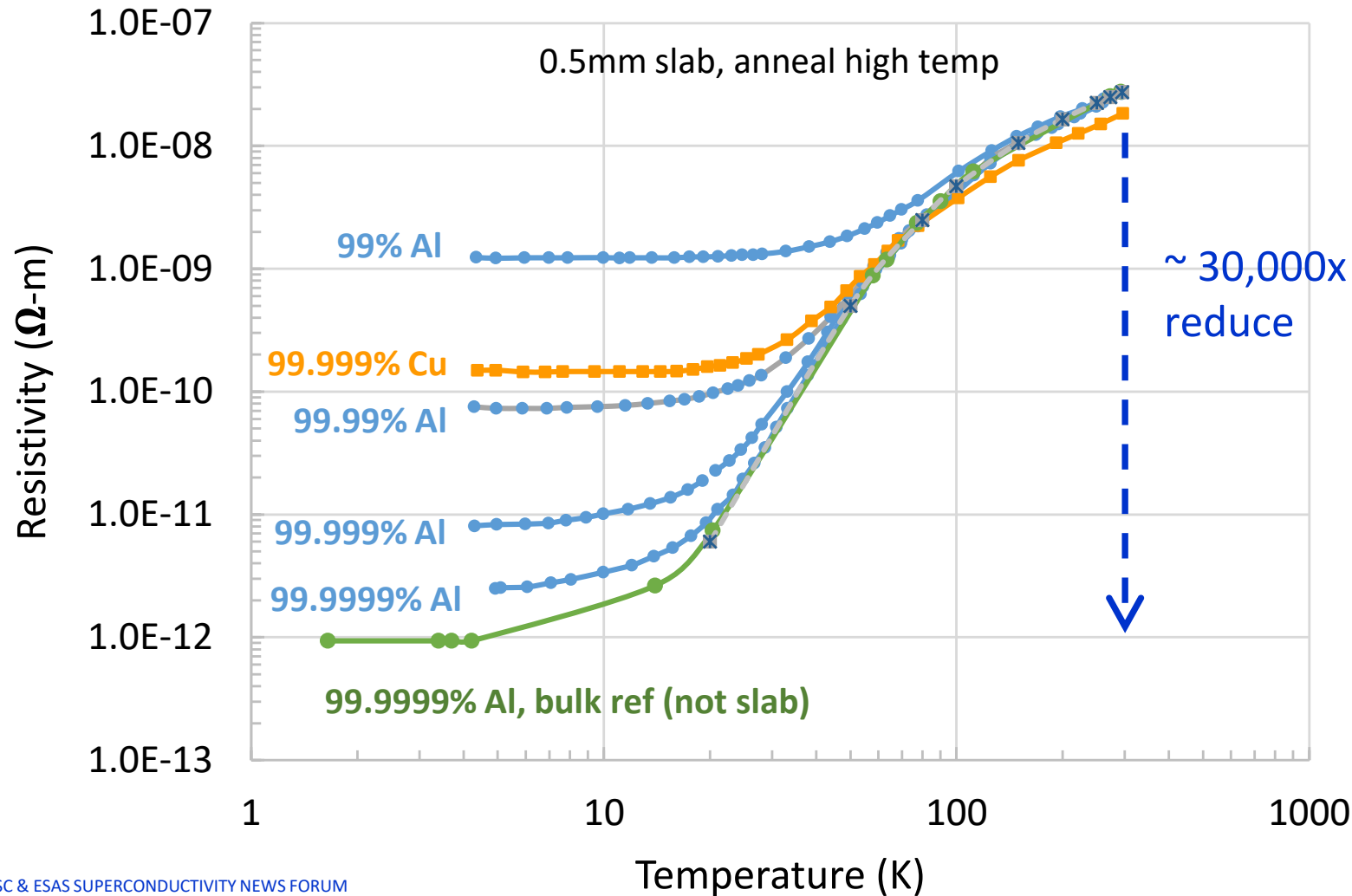
Power Distribution System, Demo: 1 MW, 1kV, 1kA

1 MW, 1kV, 1 kA Demo	Year 1	Year 2	Year 3
Hydrogen fluid flow modeling	X		
Cable Design	X	X	
Demo Design		X	
High Voltage Test, Cable			X
Purchase, Build, Demo		X	X
System Integrate			X



Ultrapure Al 99.999+% wire for Aerospace

lowest resistivity material that exists



Al,Cu Resistivity (ρ)

Cu $\rho_{300K} = 1.68 \cdot E-08$ (Sumitomo)

Al $\rho_{300K} = 2,74 \cdot E-08$ (Sumitomo)

Ohm's Law Heating of Wires

$$Q_{\text{heat}} = R \cdot I^2$$

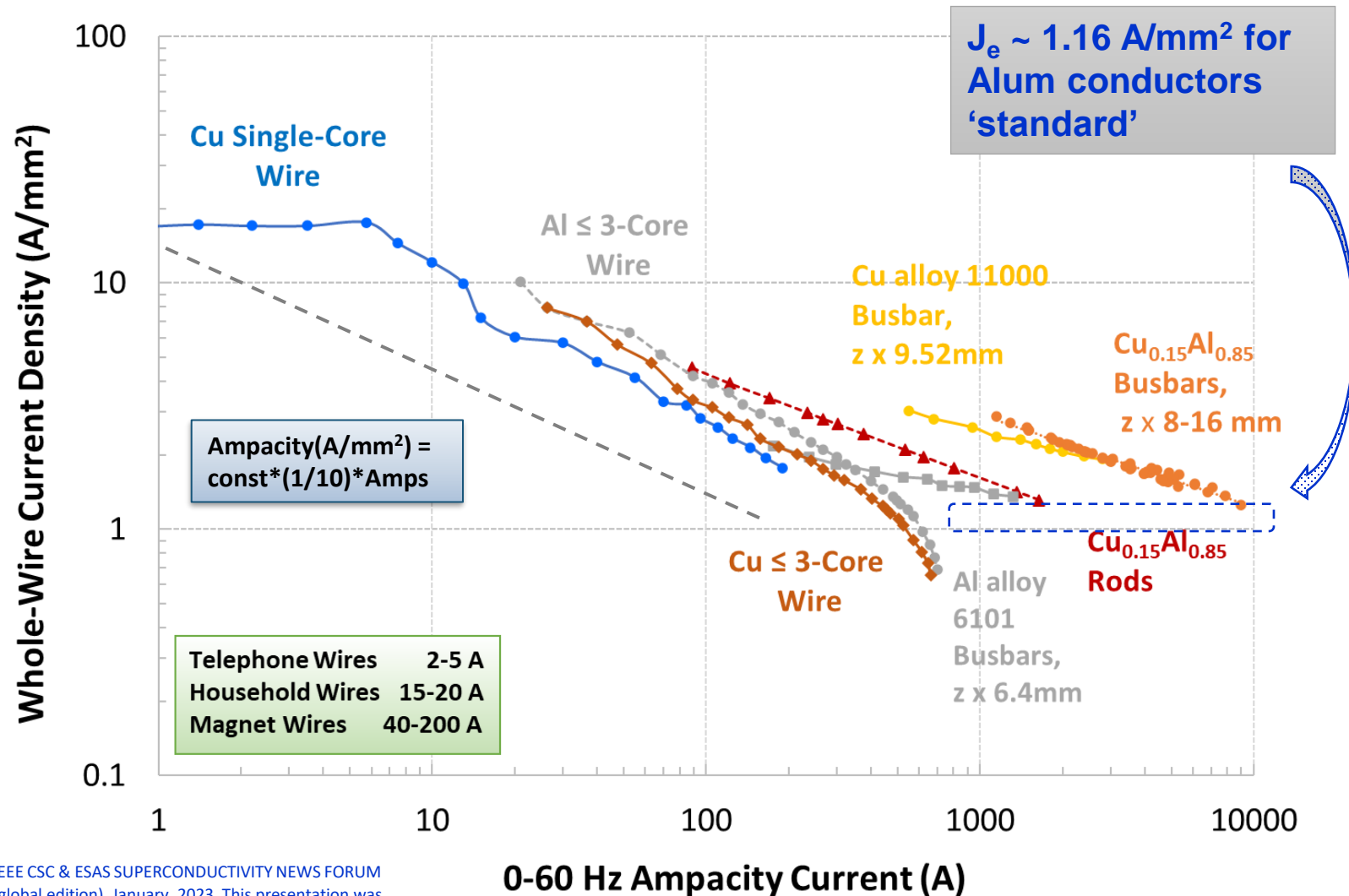
$$Q_{\text{heat}} = (\rho / \text{Area}) \cdot L \cdot I^2$$

To maintain $Q_{\text{heat}} = \text{const}$, for
 $RRR \downarrow 30,000x$, then
Wire Area (A) $\downarrow 30,000x$!!

1. Sumitomo Report "Refining Technology and Low Temperature Properties for High Purity Aluminum" 2013
2. L. A. Hall, NBS Technical Note 365 "Al-bulk"

High Ampacity Metal Wire/Busbar Options

Ampacity Rating (A/mm²), 50°C Rise from ~ 294K



Conditions

- Air-cooled
- Ambient = 20-35°C
- Temp Rise = 40-50°C
- Amp values provided by companies, based on NEC rules

Ampacity Definition:

Controlled by Ohms Law:

$$Q_{loss} = R * I^2,$$

$$R = \rho * \frac{L}{A}$$

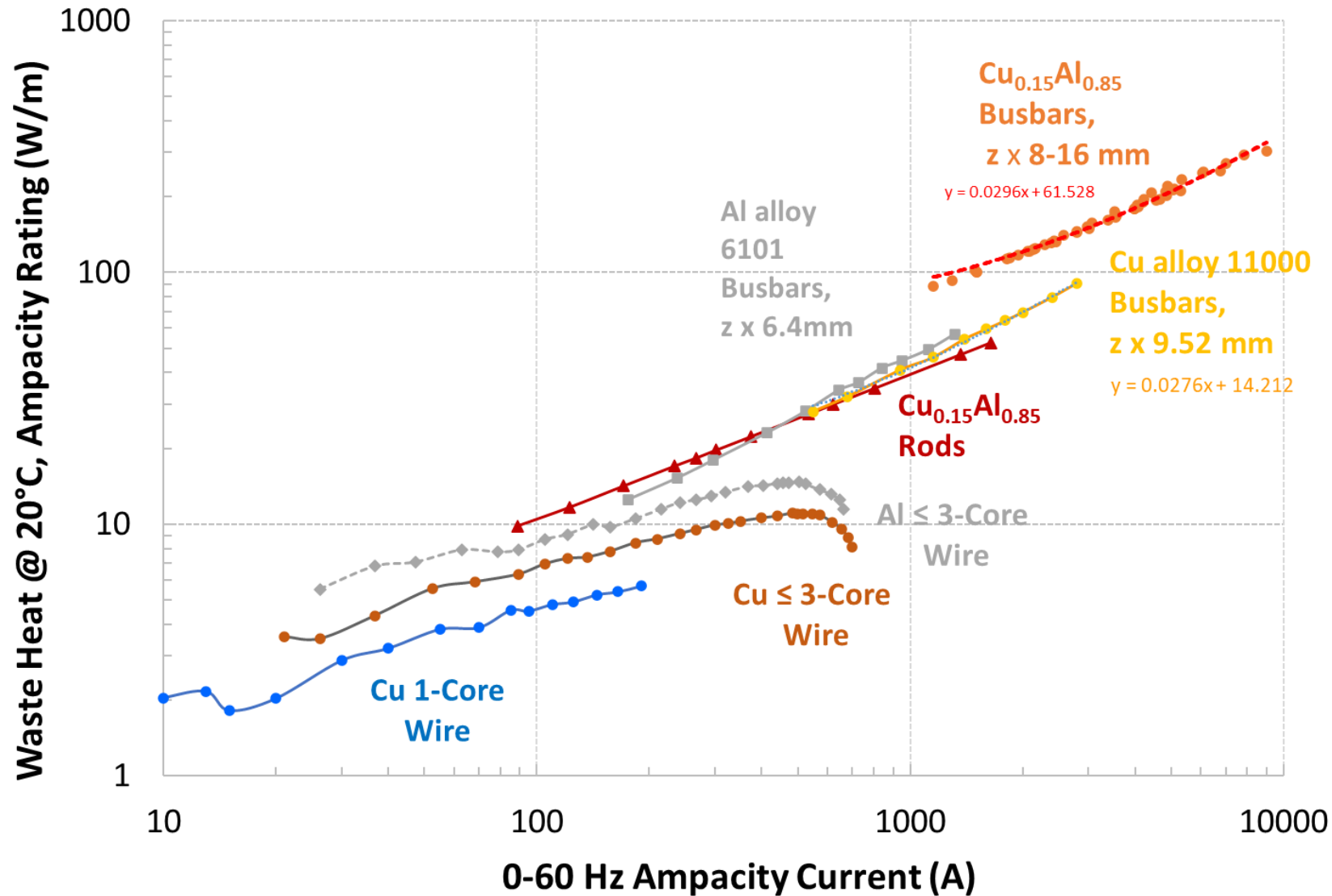
Therefore:

$$J_e = \frac{Q_{loss}}{\rho * I_c}$$

with Q_{loss} and $\rho \sim const.$ J_e is the current level at which Ohmic heat loss can be managed, to prevent thermal runaway; depends on time-duration; e.g. steady-state, pulsed, etc..

High Ampacity Metal Wire/Busbar Options

Waste Heat (W/m), 50°C Rise from ~ 294K

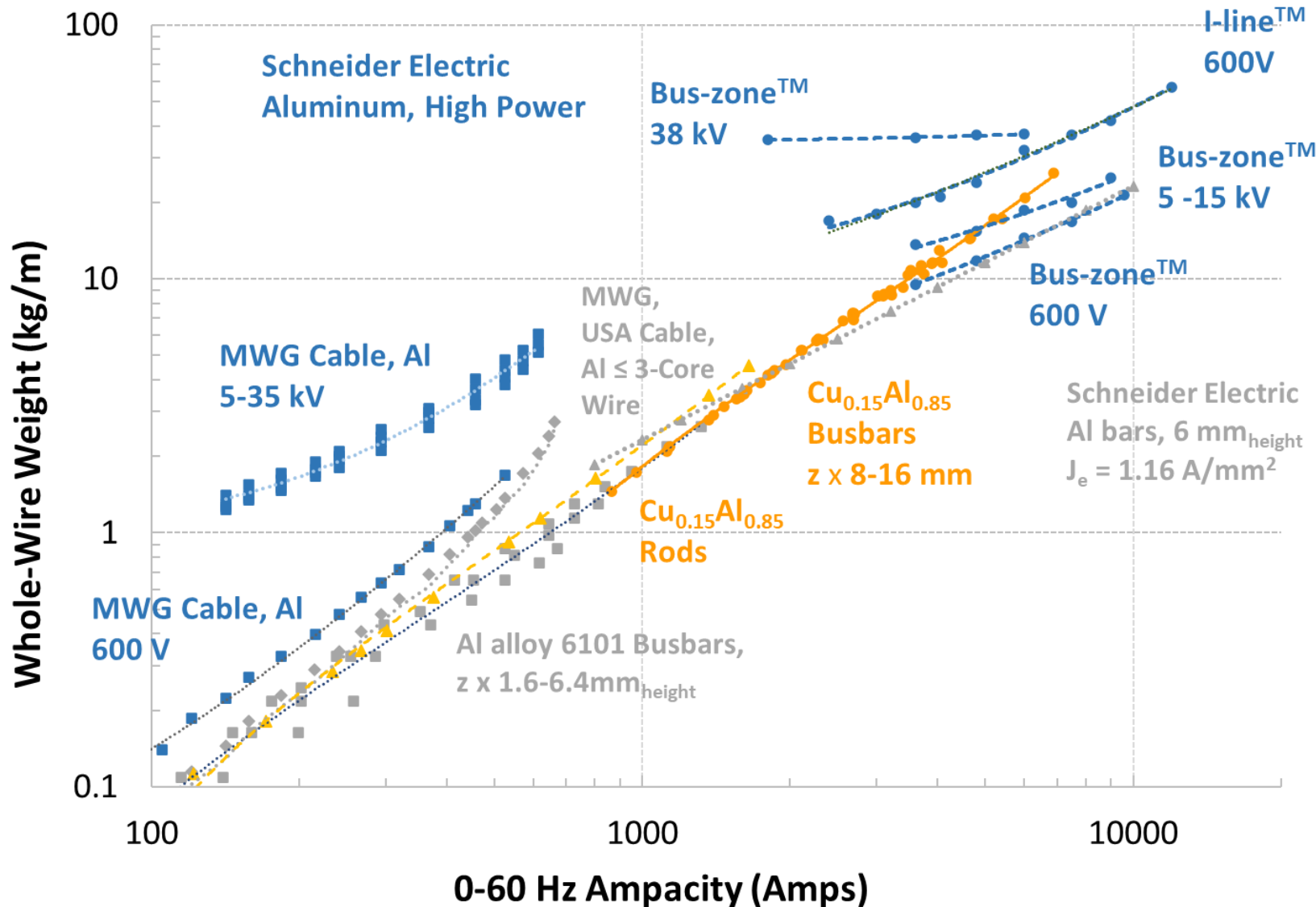


Conditions

- Insulated wires
- Ambient = 20-35°C
- Temp Rise = 40-50°C
- Amp values provided by companies, based on NEC rules

High Power Al Busbars, with 0.6-38 kV conduits

Mass Density (kg/m), 50°C Rise from ~ 294K



Conditions

- Insulated wires
- Ambient = 20-35°C
- Temp Rise = 40-50°C
- Amp values provided by companies, based on NEC rules

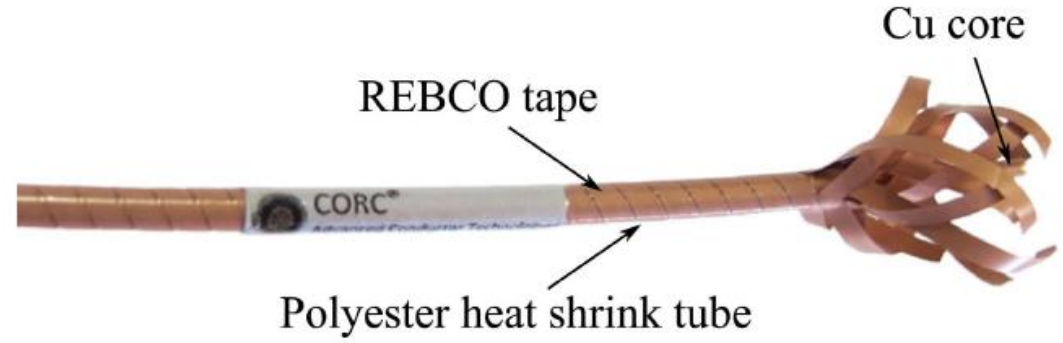
- I-line™ 600V is ~2x heavier than Bus-zone™; 600V, much easier and lower cost connections

- Must compare full systems, and not just busbars

HTS Cable, Y-Ba-Cu-O (YBCO) Conductor-Round-Core (CORC™)

Type	HTS Cable Mass	J_e @ 4.2K	I_c @ 4 K	I_c @ 20 K	I_c @ 77 K	HTS AC Loss	Reference
	(kg/m)	(A/mm ²)	(A)	(A)	(A)	(W)	
CORC Cable C0b	0.036	1,198	12,402	~ 8,900	1,560	?	Wang X, Supercond. Sci. Technol. 31 (2018) 045007 (10pp)

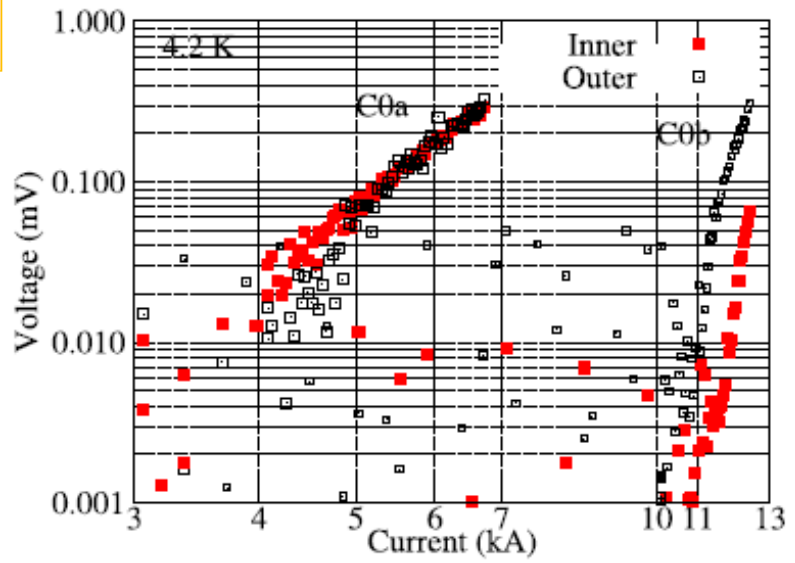
Advanced Conductor Technologies, LLC



**Weight = 0.036 kg/m, for $I_c(20K) = 8,900$ A
- Mass includes Al 99.999% former**

Table 1. Parameters of the two types of CORC® wires used for the three-turn CCT magnets C0a and C0b.

	16-tape wire	29-tape wire
Magnet name	—	C0a
Wire diameter	mm	3.09
Diameter of Cu core	mm	2.34
Number of tapes	—	16
Tape width	mm	2
Cu plating thickness	μ m	5
Substrate thickness	μ m	30
Cross sectional area	mm ²	7.5
Percentage of Cu area	—	62%
Wire length	m	2.3

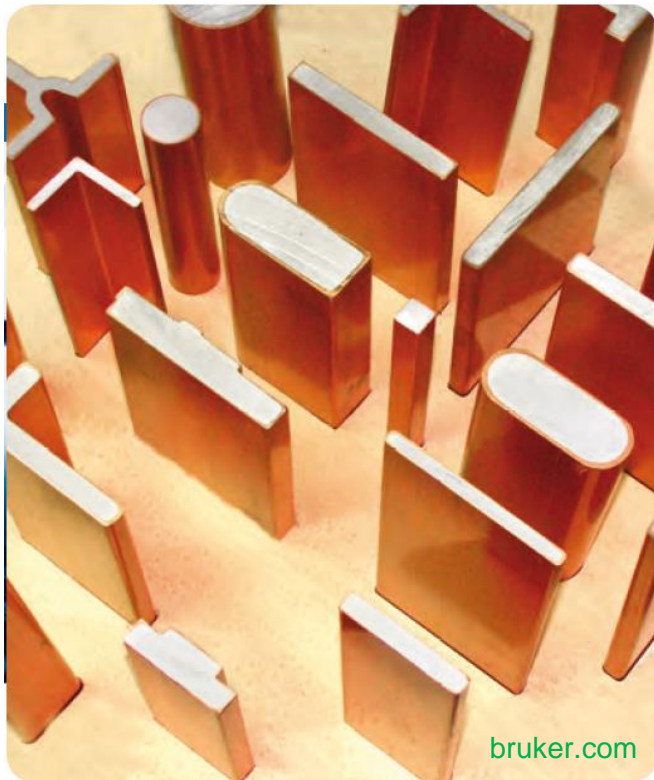


C0a, $I_c(4.2K, sf) = 6,748$ A
C0b, $I_c(4.2K, sf) = 12,402$ A
~ 10x higher than 77K, sf

Figure 9. Layer voltage as a function of current for both magnets at 4.2 K, self-field, log-log scale.

Metal Busbar Options for ~ 8,000 Amps

Liquid Cooling and Al 99.999% @ 20K



bruker.com

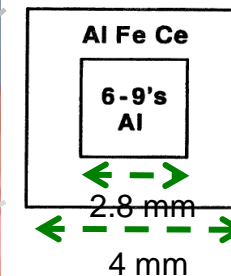
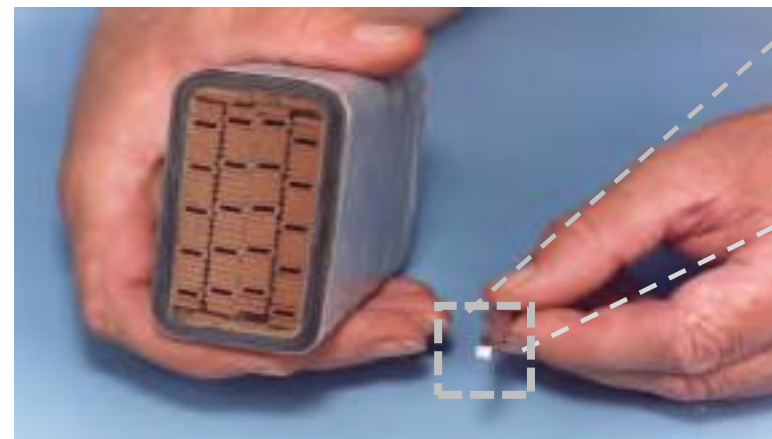
$\text{Cu}_{0.15}\text{Al}_{0.85}$ Cuponal™ @ 294K

Four bars ~ 7,000 Amps

$$J_e (8\text{kA}) = 1.6 \text{ A/mm}^2$$

$$\text{Mass (8kA)} = 20.7 \text{ kg/m}$$

$$\text{Waste Heat (8kA)} = 307 \text{ W/m}$$



Water-cooled Cu @294K

$$J_e (8\text{kA}) \sim 3 \text{ A/mm}^2 @ 294\text{K}$$

Hyperconducting Aluminum @ 20K

≥ 99.9999% purity, RRR > 1000

- $J_e(20\text{K}, H=1-2\text{T}) = 450-500 \text{ A/mm}^2$
- Weight(8kA) = 0.042 kg/m
- Waste Heat (8kA) = 212 W/m
- Al = 3.3x lighter than Cu !

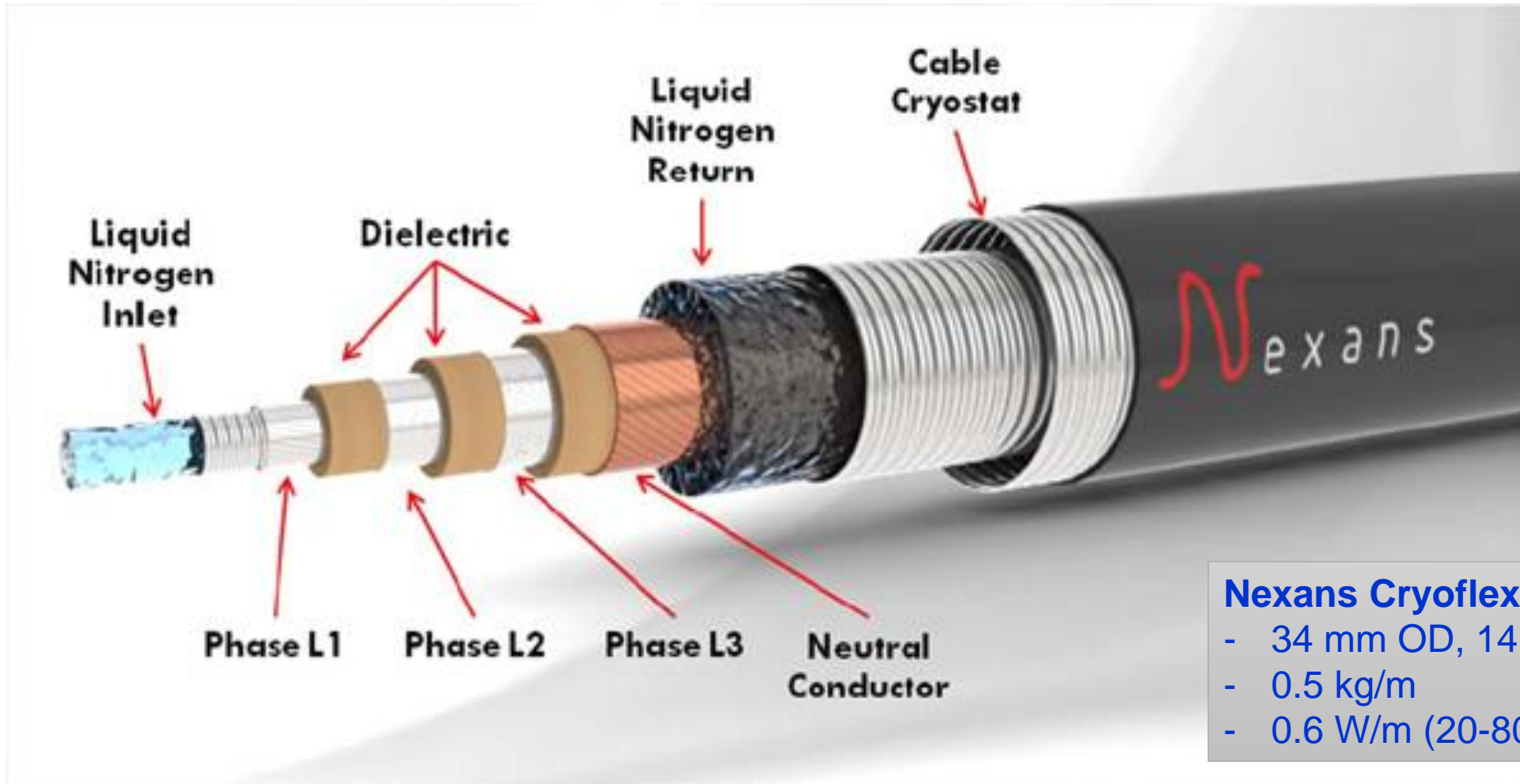
- Ref. C. Oberly (RQQM), IEEE Trans. Magn.. 27(1), 458 (1991)

For metals:

$$J_c = \frac{Q_l}{\rho * I_c}; \quad \rho_{Al} = \frac{\rho_{Cu}}{500 \dots 8,000}$$

Superconducting Cable Design, Nexans

Including Cryocooling, example Option

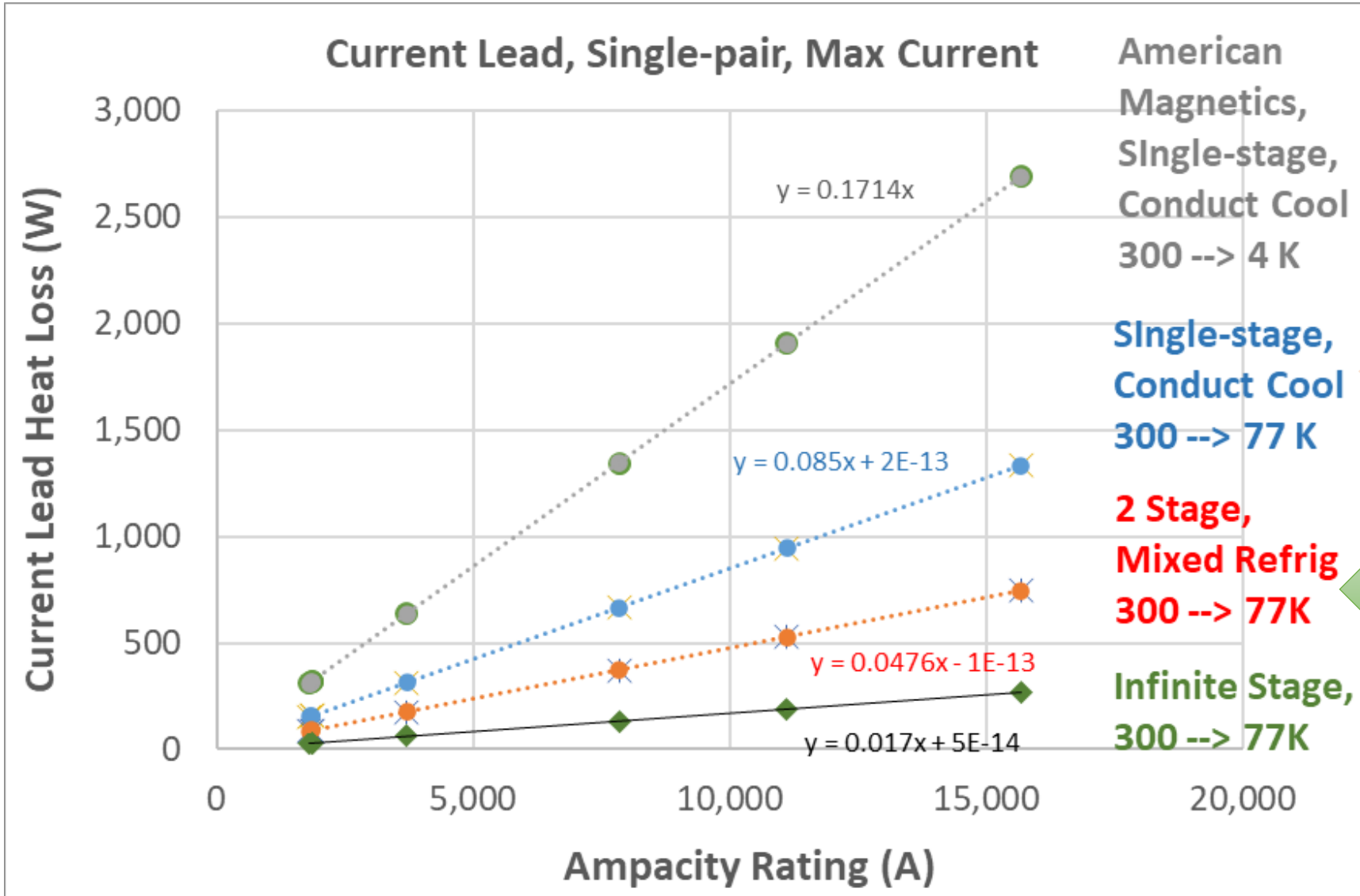


Nexans Cryoflex Tubing (COTS)

- 34 mm OD, 14 MM ID
- 0.5 kg/m
- 0.6 W/m (20-80K)

Cu Metal Current Lead Options

Waste Heat Generation (W)



American
Magnetics,
Single-stage,
Conduct Cool
300 → 4 K

Single-stage,
Conduct Cool
300 → 77 K

2 Stage,
Mixed Refrig
300 → 77K

Infinite Stage,
300 → 77K

Classic Paper. 1-stage

R.McFee, Rev Sci Instr. 30, 98 (1959)

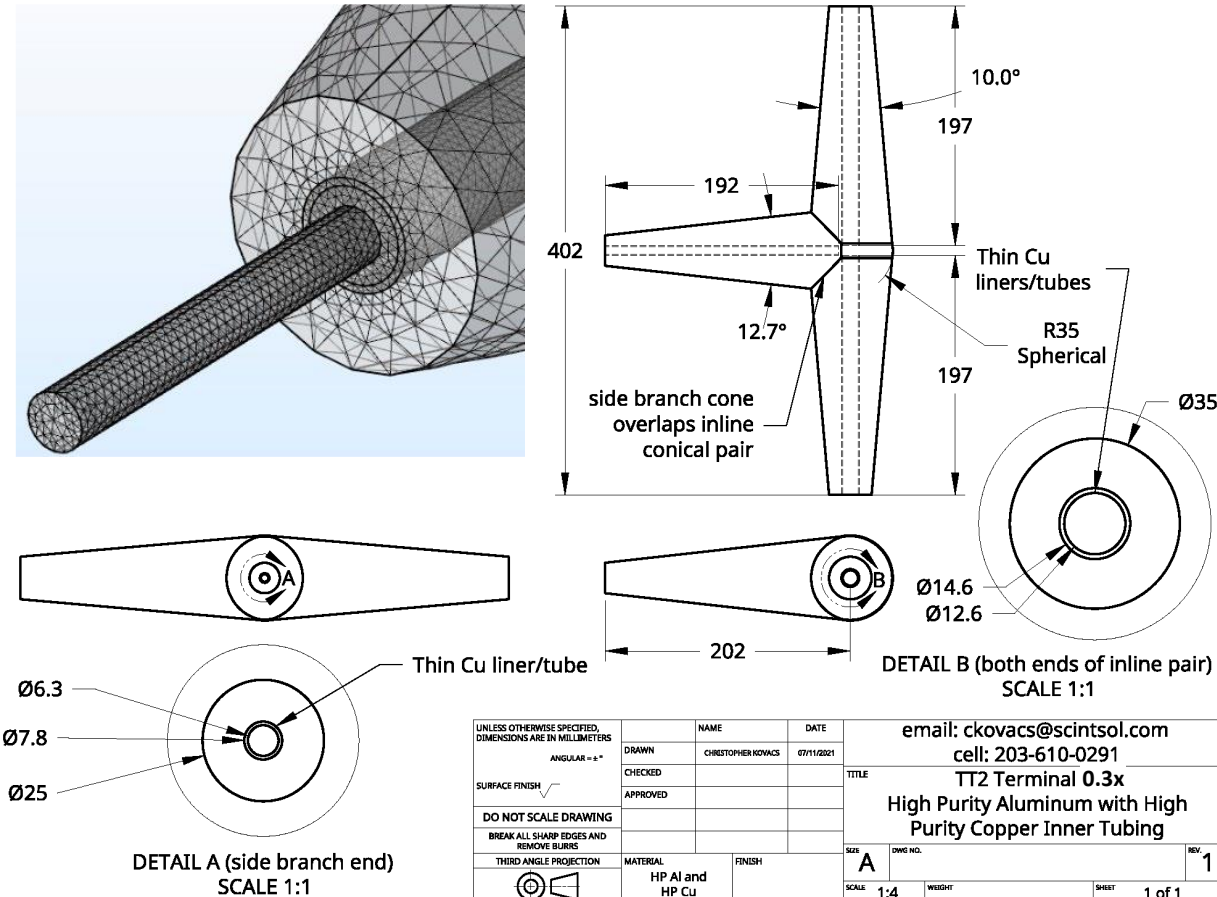
- Only needed if YBCO or BSCCO-2223 Cables.
- If MgB_2 then add YBCO or BSCCO current lead 77 K → 20 K
- Use this for spreadsheet

E. Shabagin et al, IOP Conf. Ser.: Mater. Science Eng., v502, 012136 (2019)

Note: Heat losses @ zero current (power) are ~ 60% of max current.

Multiphysics Simulation Details

- COMSOL 5.3.0.223
- Heat transfer, AC/DC, structural mechanics modules
- Electromagnetic heating, magnetic fields, electric currents, thermal stress interfaces
- Cryogenic Young's mod, Poisson ratio, coefficient of thermal expansion, electric resistivity, magnetoresistance, thermal conductivity found in literature
- Elastic regime (no plastic deformation)
- Volume integrated joule heating and density to find total power and mass
- Fixed surface temperatures 30, 40, 50 K
- Fixed contact resistivity 0, 20, 200 $\text{n}\Omega \cdot \text{cm}^2$
- Three different sizes 1x, $\sim 0.6x$, and $\sim 0.3x$ mass



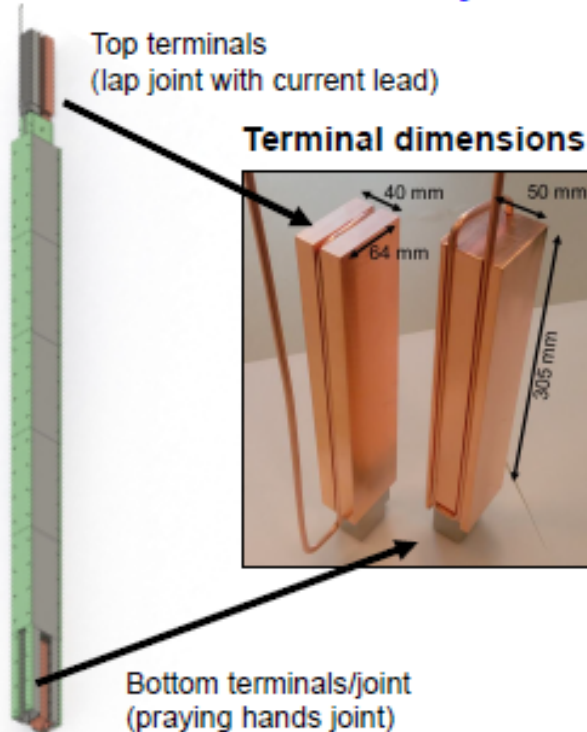
Y-Ba-Cu-O Conductor-on-Round-Core (CORC[®]) Mechanical Connector Joints

6x1 CORC[®] CICC terminations and joints

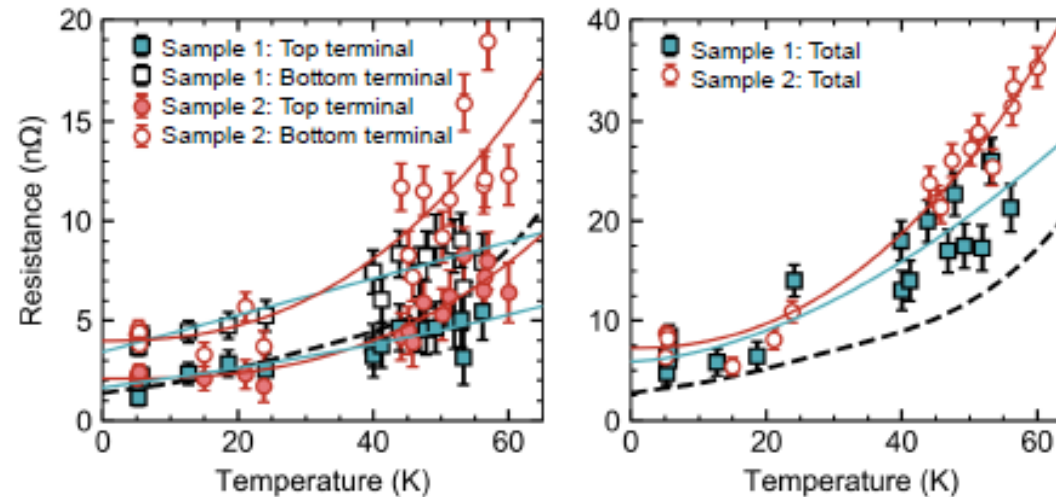
$$Q_{\text{loss}} = R \times I^2$$

$$Q_{\text{loss}}(20\text{K}, 10\text{kA}) \sim 1 \text{ W}$$

Two samples tested in series at the SULTAN test facility



6x1 CICC terminations: $R(5 \text{ K}) = 2 \text{ to } 4 \text{ n}\Omega$
Splice between two samples: $R(5 \text{ K}) = 3.3 \text{ n}\Omega$



Tim Mulder's thesis: *Advancing ReBCO-CORC Wire and Cable-In-Conduit Conductor Technology for Superconducting Magnets*
<https://doi.org/10.3990/1.9789036546164>

CORC “Tee” joint distribution:

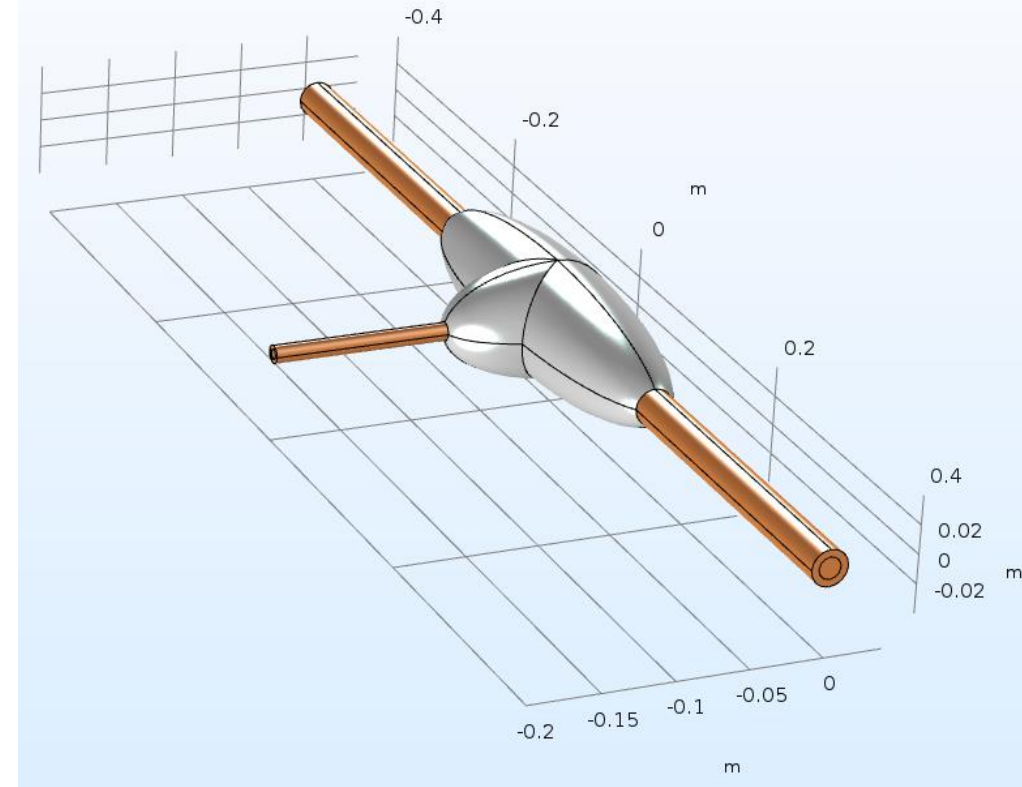
1 kV, 20 kA Main-Line \rightarrow 2.5 kA Motors

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- To greatly reduce mass, use tapered and pre-tinned CORC cables and a staged T-shaped terminal.
- A bulge will be required near the junction to enhance current injection from the main cable to the 2.5 kA branch to a motor.
- Entire T-terminal can be made of Cu but more than 50% mass can be saved if the bulge volume is made of Al
- Bonding between Al/Cu is performed with shear and hydrostatic pressure

• **Heat Loss = 2-4 W @ 30K, 20 n Ω -m² (!)**

• **Mass = 2.8 – 6.2 kg, varying with temp and power loss choice (!)**



Lapovok “Bimetallic Copper-Aluminum tube by sever plastic deformation” (2012)

C. Kovacs, et al, *J. Phys. Conf. Series* **1975**,
012036 (2021)

Power Distribution Components

Parameterize Mass, Heat Loss Scaling for 0 → 20,000 Amps

Battery, Fuel-Cell → Motor

Pink = large mass, heat loss

Power Transmission Cable Component		Mass Parameterization	Heat Loss Parameterization	Reference, Comments
Metal Cable ~ 300K, single	Cu _{0.2} Al _{0.8} Cuponal™	Mass(kg/m) = 0.0027*Wire(A)	Heat-Loss(W/m) = 0.0437*Wire(A)+45.14	www.brucker.com , ampacity tables
Metal Cable Conduit		don't have yet		
Current Lead 300K --> 77K	Cu _{0.2} Al _{0.8} Cuponal pair™	Mass(kg) = 0.0062*Wire(A)-3.8606		1st estimate, Cuponal current lead
Current Lead 300K --> 77K	Cu, pair	mass not provided	Heat-Loss(W) = 0.085*Wire(A)	1-stage, R.McFee, Rev Sci Instr. 30 , 98 (1959), asked for mass Cu/Bronze
Supercon Lead 77K --> 20K	BSCCO ?, pair	Mass(kg) = 0.00066 *Wire(A)	Heat-Loss(W) = 0.00034*Wire(A)	HTS 110, private communication
Metal to Supercond Joint	Cu to YBCO, pair	~ zero, cable plugs insert into lead	Heat-Loss ~3-10W, 5-20 nOhm@20K, 15kA	Adv Cond Tech., EUCAS 2019
HTS Cable, single, "to"	YBCO CORC™	Mass(kg/m) = 4.0438 E-6 *Wire(A)	DC loss = 0, AC Loss Transients?	X. Wang, SuST 31 , 045007 (2018), info..
HTS Cable, single 'from'	YBCO CORC™	Mass(kg/m) = 4.0438 E-6 *Wire(A)	DC loss = 0, AC Loss Transients?	" 8.9kA@20K, 3.63mm OD, 2.56 mm ID; Al
Hypercond Al Cable "to"	HP Al	Mass(kg/m) = 1.94 E-5 *Wire(A)	Heat-Loss(W/m) = 0.007155*Wire(A)	C. Oberly, IEEE Tr. Magn. 27(1) , 458 (1991)
Hypercond Al Cable "from"	HP Al	Mass(kg/m) = 1.94 E-5 *Wire(A)	Heat-Loss(W/m) = 0.007155*Wire(A)	C. Oberly, IEEE Tr. Magn. 27(1) , 458 (1991)
T-Connector, HTS to Metal	CORC™ to Cu/Al	Mass(kg) = 0.0006*Wire(A)	Heat-Loss(W) = 0.00123*Wire(A)	AFRL Calc @30K; 50K is ~4x higher
Cryoflex Tubing	OD=34 mm, ID= 14 mm	Mass = 0.50 kg/m	Heat-Loss = 0.5-0.63 W/m @ 77K --> 20K	Nexans COTS , other options possible

20,000 Amps for 1 kV = 20 MW Power; for one wing

Power Distribution Cable, Excel Calculator, V02

Outputs: Mass, Heat Loss of Components/System

	Power Transmission Cable Component		Bus Power (kW)	Bus Voltage (V)	Bus Current (A)	Piece Length (m)	Piece Mass (kg/m)	Piece Heat (W/m)	Piece Mass (kg)	Piece Heat (W)
Depends on Length	Metal Cable ~ 300K, single "to"	Cu _{0.2} Al _{0.8} Cuponal™	40000	1000	40,000	43.31	108.0	1,793	4,677	77,661
	Metal Cable ~ 300K, single "from"	Cu _{0.2} Al _{0.8} Cuponal™	40000	1000	40,000	43.31	108.0	1,793	4,677	77,661
	Metal Cable Conduit, pair		40000	1000	40,000	43.31				
	HTS Cable, single, "to"	YBCO CORC™	40000	1000	40,000	43.31	0.16	0	7	0
	HTS Cable, single 'from'	YBCO CORC™	40000	1000	40,000	43.31	0.16	0	7	0
	Hypercond Al Cable "to"	HP Al	40000	1000	40,000	43.31	0.78	286	34	12,395
	Hypercond Al Cable "from"	HP Al	40000	1000	40,000	43.31	0.78	286	34	12,395
	Cryoflex Tubing	OD=34 mm, ID= 14 mm	40000	1000	40,000	43.31	0.50	1	22	27
			Subtotals				218.4	4,159	9,458	180,140
						# Pieces	kg/piece	W/piece		
Fixed (pieces)	Current Lead 300K --> 77K	Cu _{0.2} Al _{0.8} Cuponal pair™	40000	1000	40,000	1			244.1	
	Current Lead 300K --> 77K	Cu, pair	40000	1000	40,000	1				3,400
	Supercon Lead 77K --> 20K	BSCCO ?, pair	40000	1000	40,000	1			26.40	24.00
	Metal to HTS Joint	Cu to YBCO CORC™, pair	40000	1000	40,000	1			0.00	24.00
	T-Connector, HTS to metal	CORC™ to Al/Cu @30K	2500	1000	2,500	32	1.600	3.075	51.20	98.40
			Totals						9,355	155,322
									411	28,364
									357	3,574

CHEETA V02
40 MW, 1 kV, 40 kA, 43.3m

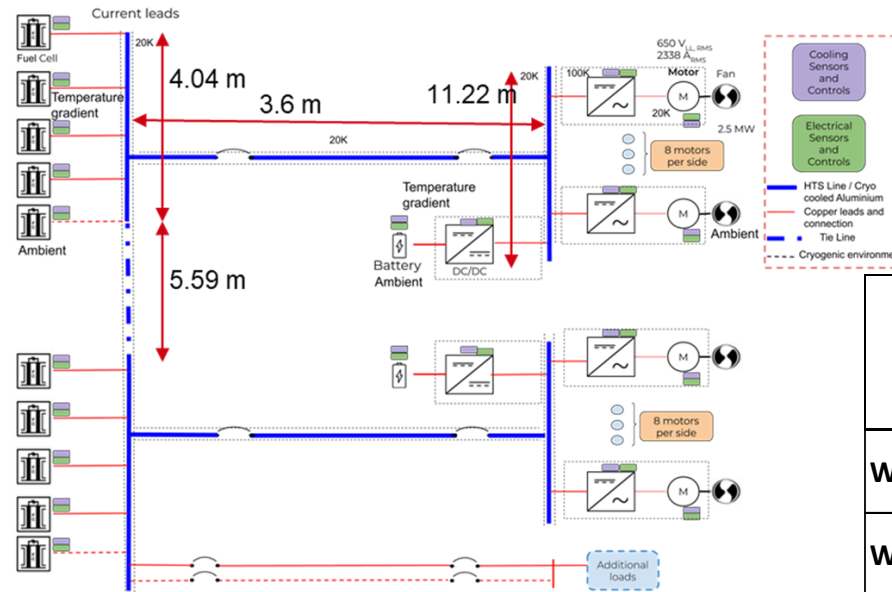
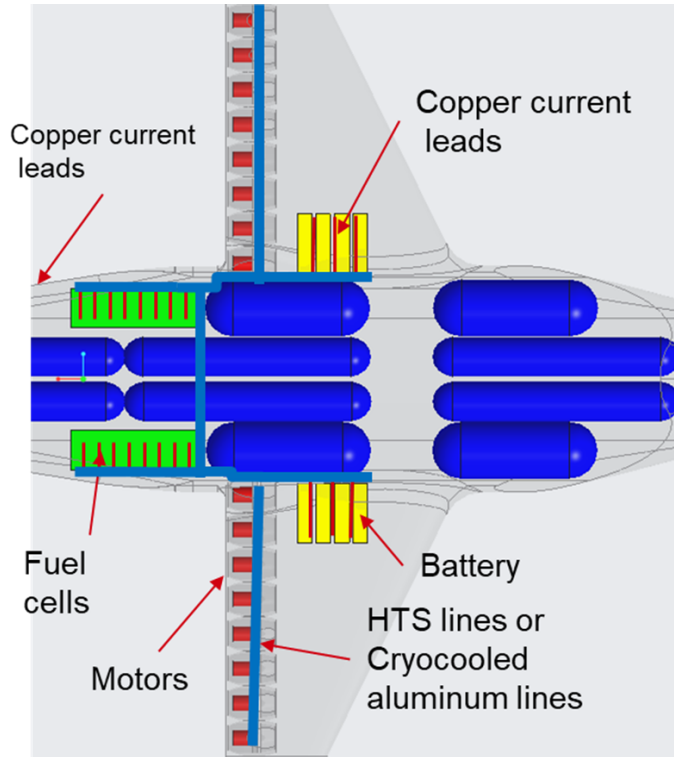
Almost 'zero' mass

>70% of supercond/cryo system

Heavy, Hot!

Power Distribution System - Mass, Heat Summary

CHEETA Liquid H₂ Fuel Cell, 40 MW, 1 kV, 40 kA, 43 m length



3 Technology Solutions

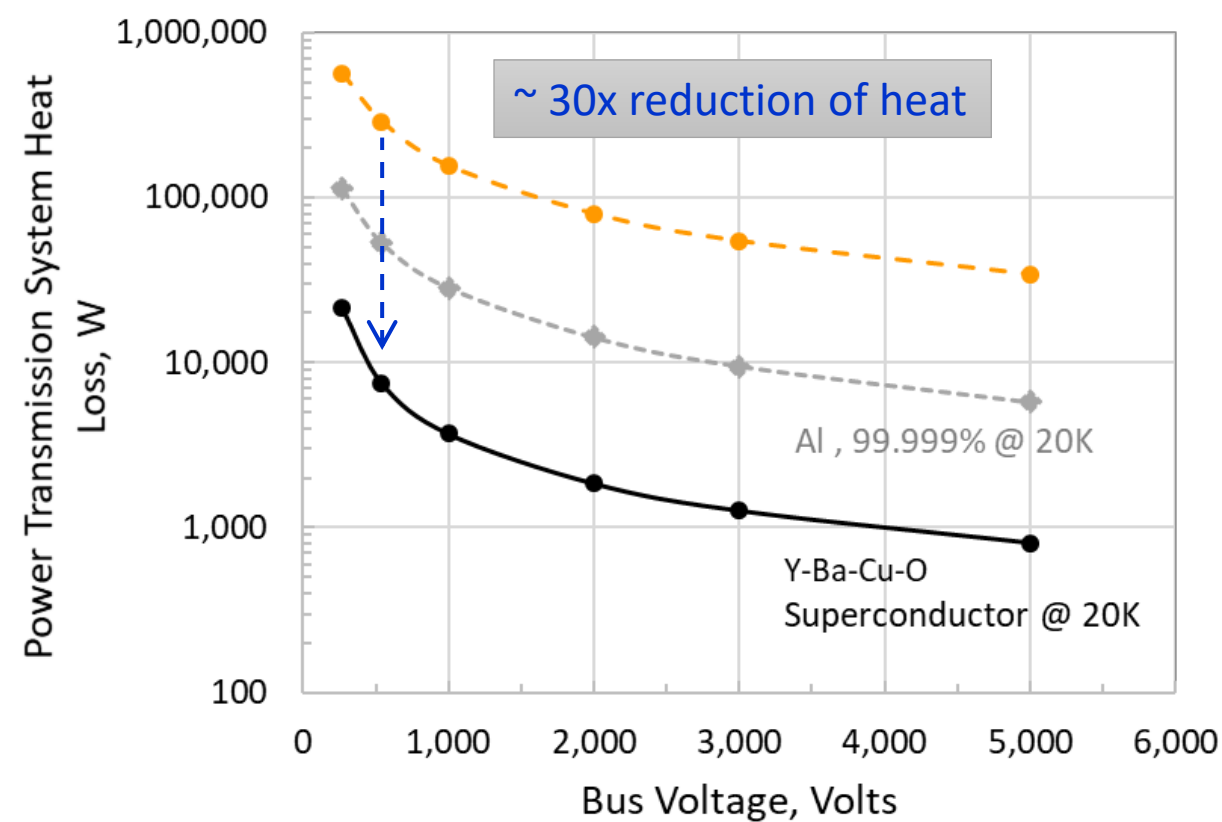
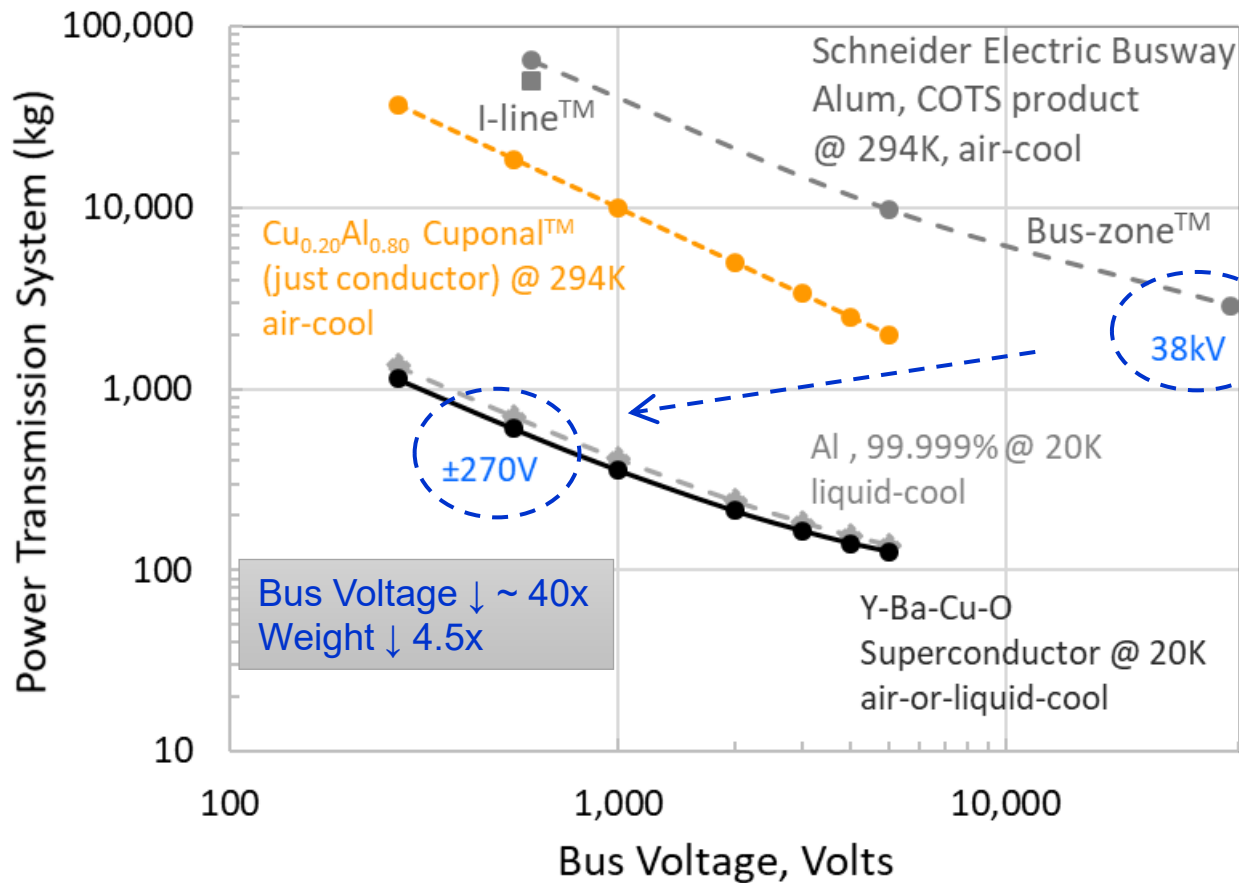
	$\text{Cu}_{0.15}\text{Al}_{0.85}$ Cuponal™ @ 294K	Al > 99.999% Hyperconductor @ 20K	YBaCuO or MgB ₂ Superconductor @ 20-65K
Weight	9,355 kg (heavy!)	411 kg (light)	357 kg (light)
Waste Heat	155 kW (hot!)	28.5 kW (hot)	3.7 kW (cool!)
Cost	medium	high	high
Complexity	low	medium	medium
TRL Level	9	4	4, aircraft 9, CERN
Protection Risks	high	high	medium, (FCL intrinsic)



Power Distribution System - Mass, Heat Summary

CHEETA Liquid H₂ Fuel Cell, 40 MW, 40 kA, 43.3 m length

3 Technology Options



± 270 V Bus is 'feasible', and 4.5x lower mass than 38kV Bus-zones™, high voltage busway is ground not air-ready

Benefits of Superconductivity/Cryogenics for CHEETA V02 Drivetrain System, *~ 8x lower mass, ~ 7x lower heat loads*



Boeing 737-800 Class, 178 Passenger
 NASA ULI CHEETA V02, liquid-H₂ + fuel-cells
 40 MW, 1 kV, 40 kA!, 43 m cable length
 - Consider only major components so far

CHEETA ULI Studies

	Current Progress, Near-Term				Future Investment			
	Power Density (kW/kg)	40 MW Power Mass (kg)	Efficiency	Waste Heat (kW)	Power Density (kW/kg)	40 MW Power Mass (kg)	Efficiency	Waste Heat (kW)
Generator	4	10,000	94	2400	40	1,000	99.8	80
Rectifier	20	2,000	98	800	40	1,000	99	400
Distribution	4.28	9,350	99.61	156	112.00	357	100	0
Motor Drive	20	2,000	98	800	40	1,000	99	400
Motor	13	3,077	96	1600	40	1,000	99.8	80
Thermal	4	10,000	98	800	150	267	100	0
Subtotals	1.1 kW/kg	36,400 kg	83.6%	6,500 kW	8.7 kW/kg	4,600 kg	97.6%	960 kW

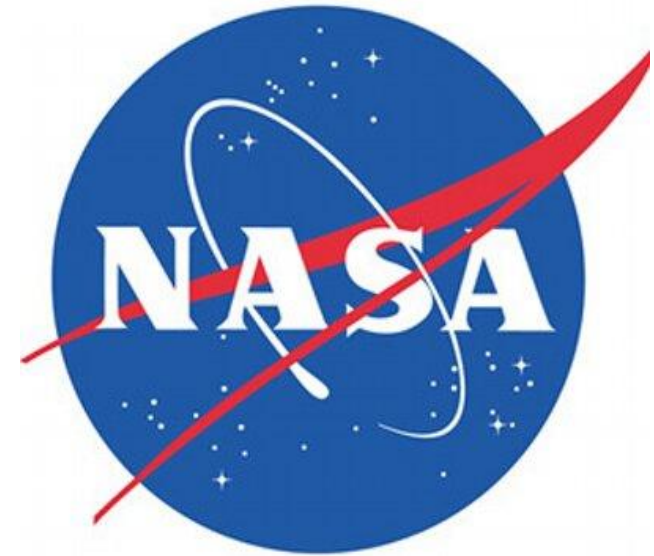
Larger than Passenger Load = 13,700 kg !! (won't work)

Summary Points, 22.5 MW EWIS Scaling Calculator

- **Superconductors arguably the best option**
 - very low heat loss, easiest to manage (gas cooling only)
 - ultralow volume and diameter, which reduces the impact of the cryoflex conduit
 - cables expensive (?)
- **Al 99.999% next best option**
 - provides large reduction of weight
 - heat loss is not insignificant, and must be cooled with liquid H₂. This adds greater complexity, and thermal management/control issues.
- **Operation at ± 270 V @ 20K is considered a viable candidate**, and is even 30x lighter than Cu, Al at 294K, 30x less waste heat
- **Scaling for all components to higher powers is not necessarily linear** - must study each component
- **Design and operation of components can be complex** -
 - trade options possible; e.g. can swap mass and waste-heat

Acknowledgements

Questions?



This work was supported by NASA under award number 80NSSC19M0125 as part of the Center for High-Efficiency Electrical Technologies for Aircraft (CHEETA)