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Next Generation Drive Train

Superconductivity for Large-Scale Wind Turbines*

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Project Objectives

The primary objective of the project was to apply low temperature superconducting technology to the design of a direct-drive wind turbine generator at the 10MW power level in order to reduce the Cost of Energy (COE).

The 6 month project focused on the design of the generator, an evaluation of the commercial viability of the design together with an identification of high risk components.

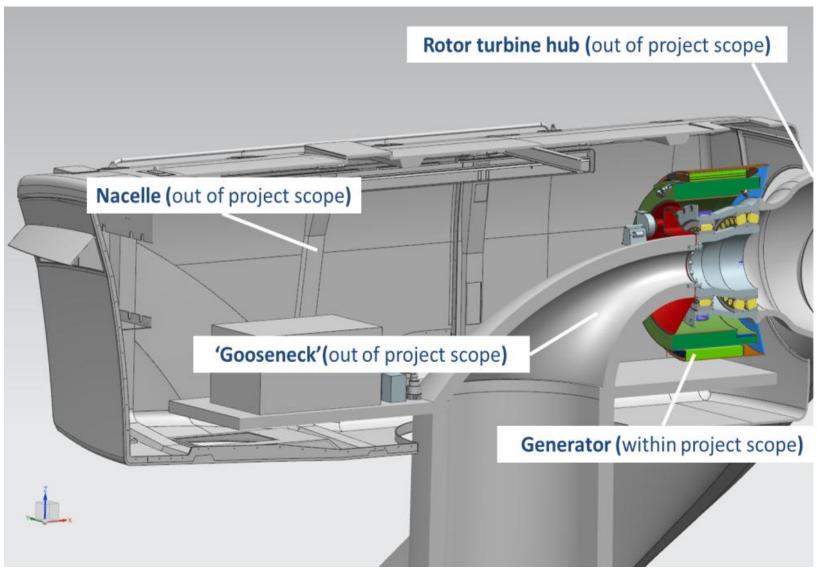


How can we get to a commercially viable product as quickly and as pragmatically as possible?

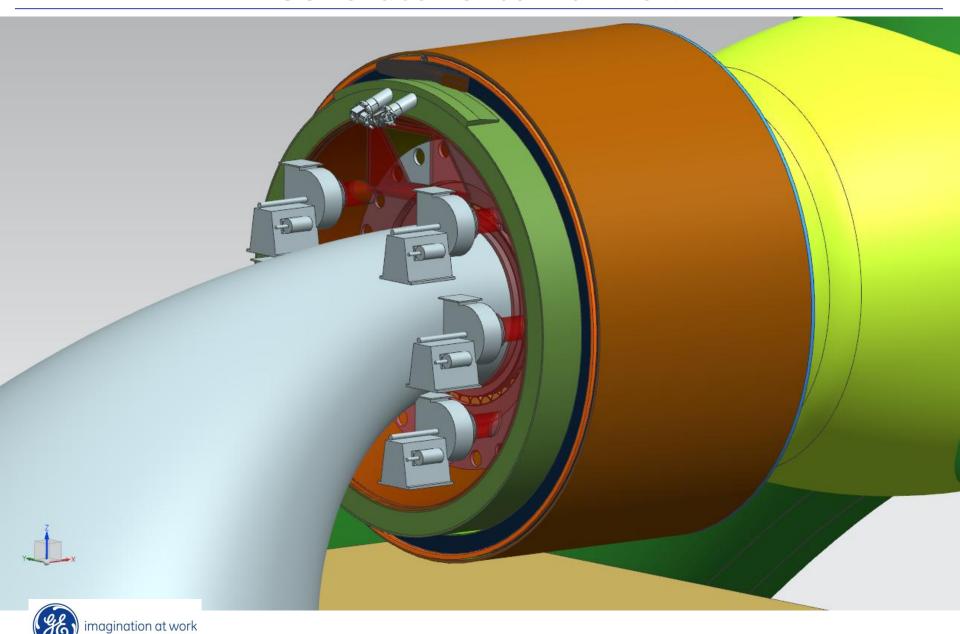
- ☐ Use readily available, cost-effective proven superconductor
 → LTS
- □ Reduce risk at the pre-design stage → e.g. eliminate the cryogen transfer coupling
- ☐ Stationary field → Utilize GE Healthcare MRI cooling technology know-how
- ☐ Utilize conventional manufacturing materials and existing production processes
- ☐ Utilize GE's extensive knowledge of wind system integration



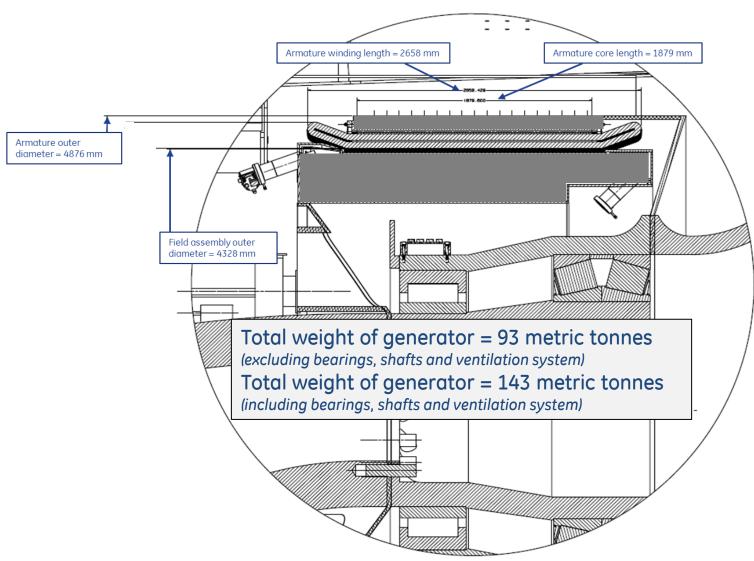
Project Scope



Generator external view

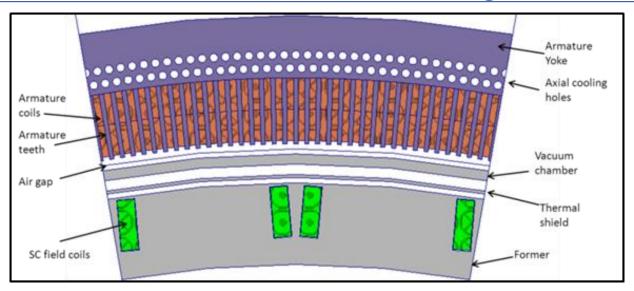


Key Generator Dimensions

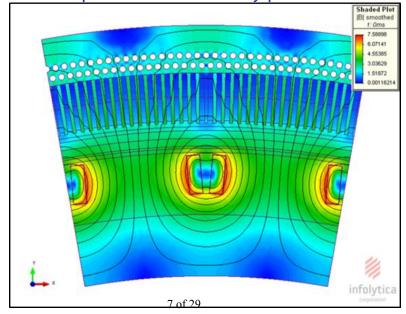




Cross-sectional view of the generator



Open circuit flux density plot (Tesla)





Generator final design parameters

Parameter	Value	
Rated Power	10 MW	
Rated Speed	10 rpm	
Rated torque	10 MNm	
Rated Voltage	3300 V line-line	
Rated Current	1750 A	
Rated Power Factor	1.0	
Full Load Efficiency	95-96%	
Physical Air gap length 19 mm		
No. of poles / No. of slots	36 / 648	
Armature Winding Type	3 phase, 2 layer, lap, form wound	
Insulation	Class F (with Class B temperature rise)	
SC Field MMF	928000 AT/pole	
Armature Cooling Axial air cooled thru air gap and y		

Figure of Merit	10 MW SC Generator	Conventional PM Generator	Increase with SC
Shear Stress	179 kPa	85 kPa	2X
Torque Density (EM only)	197 Nm/kg	94 Nm/kg	2X
Torque Density (Drivetrain)	92 Nm/kg	44 Nm/kg	2X
Peak Fault Current	15 p.u. (L-L-L)	4 p.u.	4X
Peak Fault Torque	12 p.u. (L-L)	2 p.u.	6X



Generator losses and efficiency

Generator Load Condition 10MW @ 10rpm				
Arm winding DC Loss	363 kW			
Armature AC Loss	56 kW			
Armature Yoke Loss	5.7 kW			
Armature Teeth Loss	5.6 kW			
Armature Core Clamp Loss	2.1 kW			
Field AC Loss (incl. vessels)	2.6 kW			
Armature Slip Ring Loss	4.6 kW			
Friction and Windage	negligible			
Cryocooler power (3)	22.5 kW			
Cooling Air Blowers (6)	39 kW			
Total Loss	501 kW			
Efficiency	95.0%			



Armature slip ring design

Based on GE-Hydro design and operational experience with 3 phase, 3500A, 17kV, 100MW slip ring/collector systems.

Performance factor for previous installation after 3 years of commercial operation (2008):

$$PF = \frac{\sum MW.h \ (Actual)}{MW.h \ (Scheduled)} = 99.7\%$$

(equivalent of 3.28 days outage out of a total of 1095 days)

Current Design:

- No. of slip rings = 4
- Slip ring OD = 3m
- No. of carbon brushes per slip ring = 30
- Current per brush = 60A
- Rotational speed = 10rpm
- Total operational loss @ 10MW = 4.6kW
- Air-cooled assembly
- Brush wear per year < 2 mm @ 10rpm

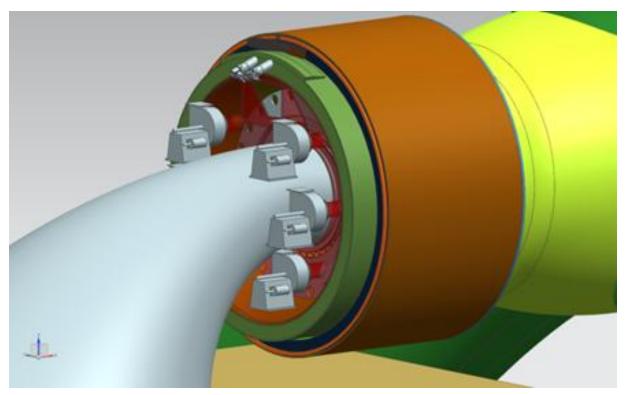








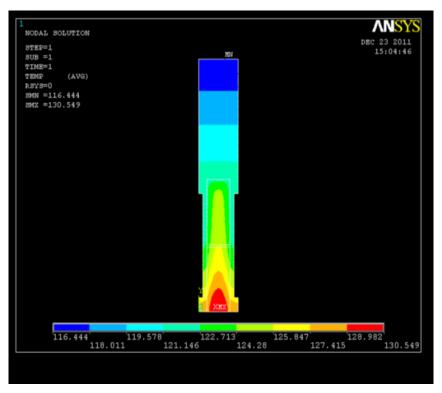
Generator cooling configuration

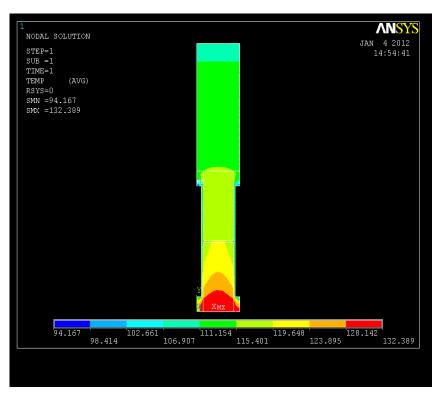


- Six air blowers are mounted to the field support plate.
- They are belt driven units with 5hp ,3600 rpm, 460 v, 3 ph, 60 Hz motors, housing drains, motor covers, shaft seal, belts and drives.
- The weight of each unit with aluminum wheel, housing and motor pedestal is approximately 251 lbs. The estimated input power for six blowers is ~39kW.



Magnetic vs Non-magnetic armature teeth





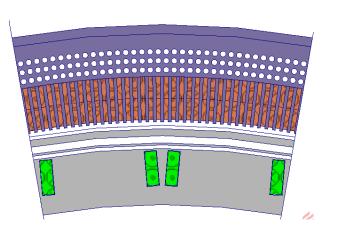
Magnetic teeth

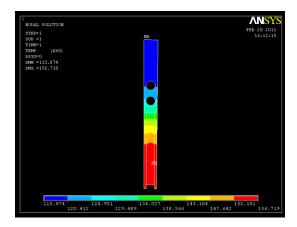
Non-magnetic teeth

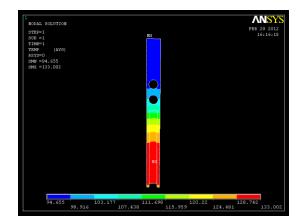
- → Hot spots in the armature and vacuum chamber wall for magnetic teeth are slightly lower than for the non-magnetic teeth design option
- \rightarrow Hot spot temperature of the vacuum wall is well below the thermal radiation limit of 80°C



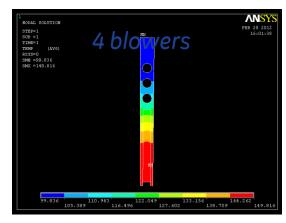
Armature cooling duct design







# of working blowers	Armature hotspot
6	133.0 °C
5	145.7 °C
4	157.2 °C



100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 100 | 10

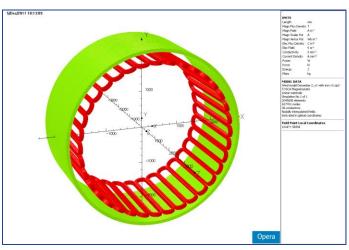
4 blowers

5 blowers

→ The final design for the cooling of the armature settled upon 2 rows of cooling holes with a total of 6 airblowers providing a more uniform air flow distribution and a measure of redundancy

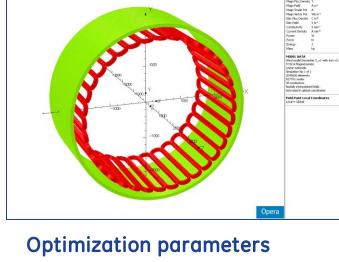


SC coil optimization

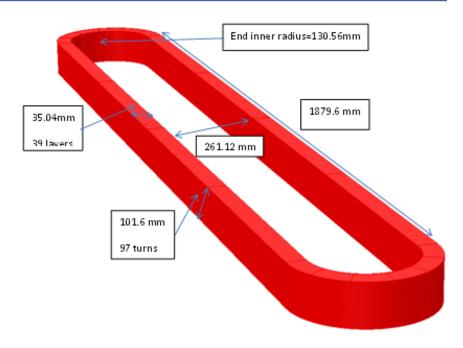




- Width of the coil
- Height of the coil
- End radius of the coil
- Operating current in the coil
- Short sample percentage for coils
- Current sharing temperature for the conductor
 - A sufficient margin is required for the stable operation of the coils. Current sharing temperature depends on the maximum field in the coil, critical current at maximum field, ratio of operating current to critical current at maximum field and ratio of maximum field in the coil to critical field of the conductor





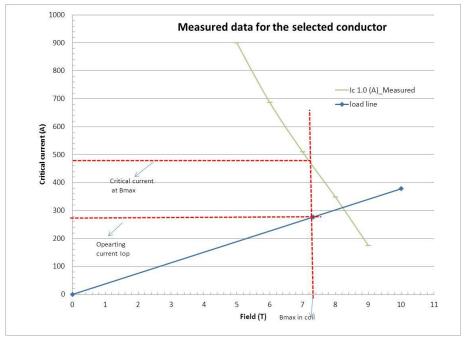


SC coil final design parameters

Parameter	Value
Coil type	Racetrack
Coil width (mm)	35.00
No. of layers in coil width	39
Coil height (mm)	101.60
No. of turns in coil height	97
Coil length straight (mm)	1879.60
Coil Inner Width (mm)	261.01
End radius (mm)	124.58
Type of conductor used	Cu-(NbTi)
Bare diameter of conductor (mm)	1.00
Insulated diameter of conductor (mm)	1.05
Operating current (Amp)	276.86
Total ampere turns (A)	928000
Maximum field in the coil (T)	7.35
Critical current at the maximum field (Amp)	466.75
Short sample percentage (%)	59.96
Critical temperature (K)	6.08
Stored energy of the system (MJ)	40.6
Inductance of all the coils (H)	1059
Total conductor used for 36coils (km)	720
Total estimated weight of the coils (kg)	3840

Parameter	Value
Type of conductor used	Cu-(NbTi)
Cu:SC	1.5
Bare diameter of conductor (mm)	1.00
Insulated diameter of conductor (mm)	1.05
Number of filaments	7400
Filament diameter (micron)	7.5





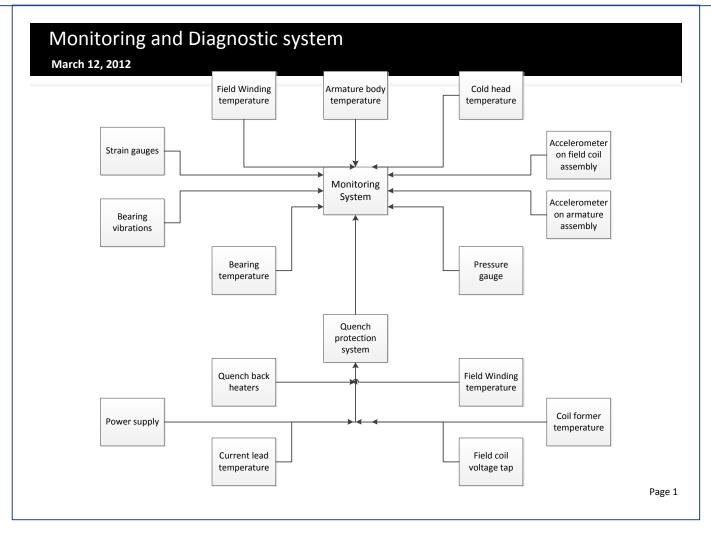
AC Losses

Loss Contribution

- 1. Loss during operation
 - a. Losses due to field current boost
 - b. Losses due to external time varying fields
 - c. Losses due to field current change
- 2. Loss during ramping
 - (i) Eddy Current loss
 - (ii) Hysteresis loss
 - (iii) Penetration loss
- Total heat loads are as follows:
 - Total heat load for single sweep= 0.17 W
 - Total heat load during operation= 0.64 W
- An Independent AC loss calculation has been performed by Dr. Robert Duckworth at the Oak Ridge National Laboratory (ORNL), based on the same assumptions provided above.
 - Total heat load for single sweep= 0.32 W
 - Total heat load during operation= 0.80 W
- → The final values of the losses are different and will need to be addressed via tests.



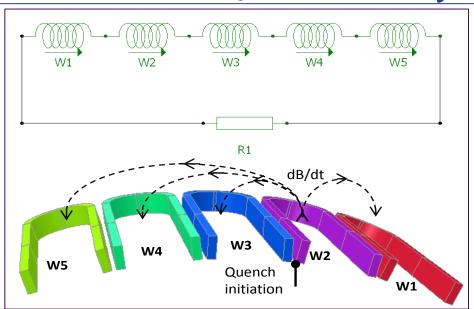
IEEE/CSC & ESAS EUROPEAN SUPERCONDUCTIVITY NEWS FORUM, No. 22, October/November 2012. Monitoring and diagnostics



Remote monitoring and diagnostics systems will play an extremely important role for these systems



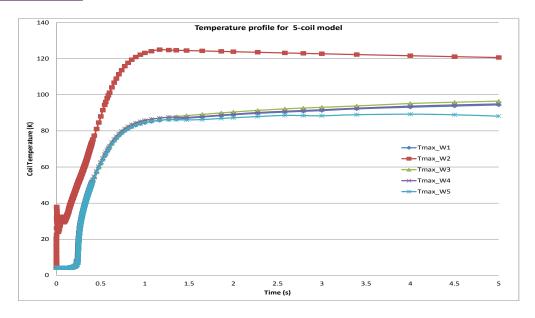
Quench Analysis – 5 Coil Model



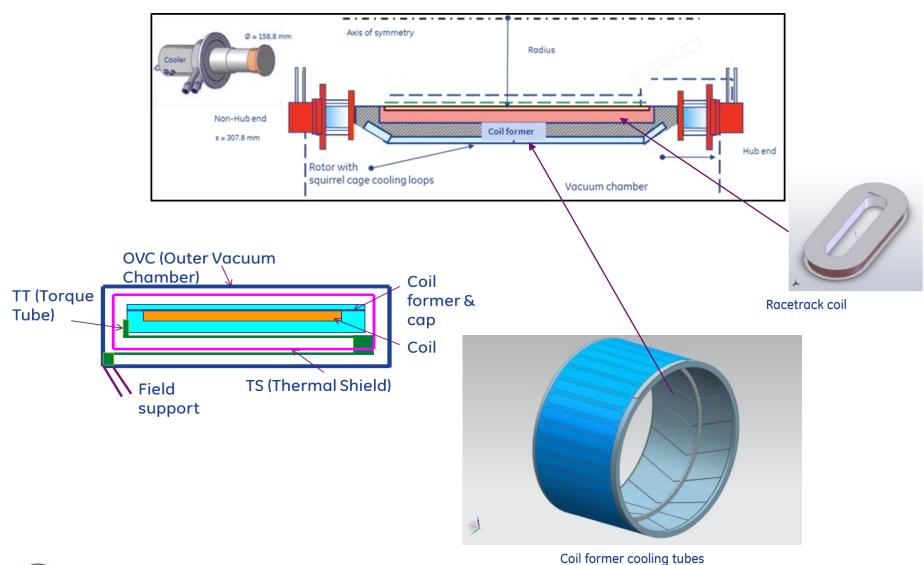
→ Coupling losses play a crucial role in the quench propagation process

- → Major portion of energy dissipated in the coil where quench initiated
- → Important factor in defining quench protection method



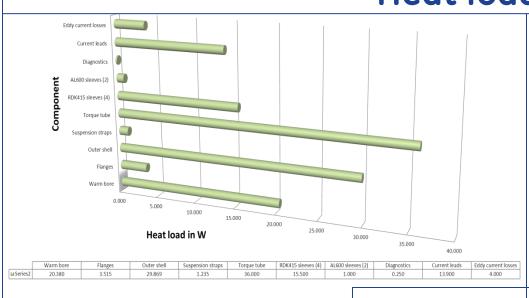


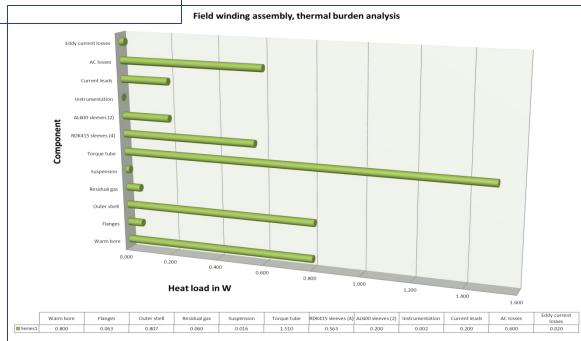
Cryogenic closed-loop cooling concept





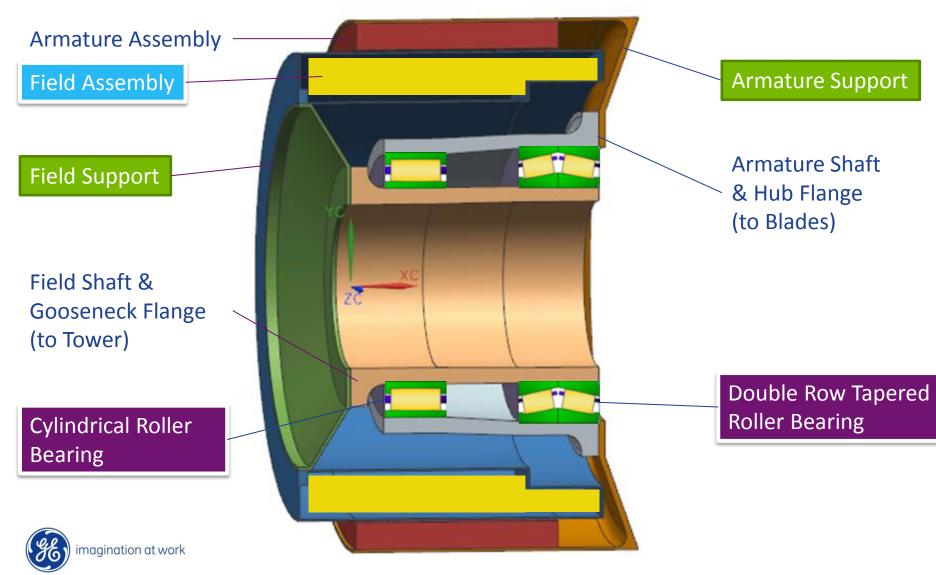
Heat loads



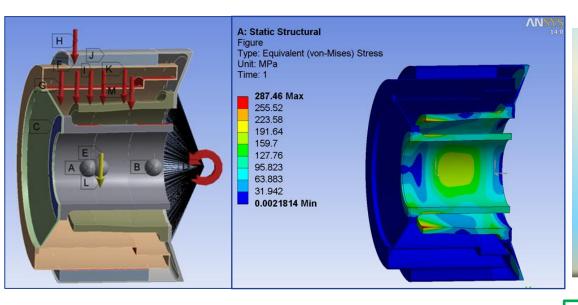




Generator mechanical sub-systems



Supporting structure analysis



- Standard Gravity
- Wind Load (Nodding Moment)
 - Nominal = 5.8e9 N-mm
 - Extreme = 5.1e10 N-mm
- EM Load
 - Nominal = 160 psi
 - Deflection Load = 5.4e6 N/in

Loads	Air Gap Closedown Spec	Air Gap Actual Closedown	Max Stress (MPa)	Safety Factor
Nominal	-	12%	37	10.3
Extreme	< 50 %	41%	287	1.3



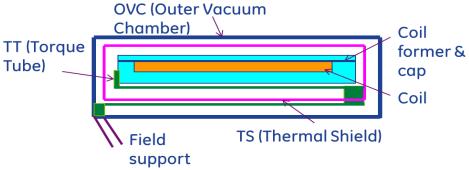
Bearing design summary

- Industry proven configuration
- Sized against wind extreme loads using Wind Turbine Design Tools
- Bearing Stiffness calculated using Bearing Design Tools
- 25+ years life estimation

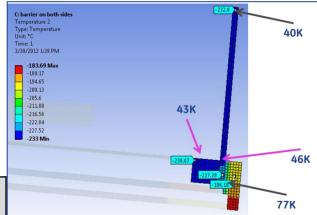
Field assembly torque tube design

The torque tube has to meet several key design constraints:

- extreme torque load conditions with respect to buckling
- exceptional fatigue properties, and in particular at low temperatures
- light weight, ease of manufacture
- minimal heat burden to magnet coil former with respect to thermal conductivity
- minimal thermal radiation
- minimum of optically black cavities or so-called "black holes"
- simple and uncompromised application of MLI should be possible



	Material	Working temperatu re (K)	Tensile Strength, Yield (Mpa)*	Max Stress (MPa)	Max Stress SF	Max Radial displacement (mm)	Max Rotational displacement (mm)
0)/0	2004 70	40.000	000			0.00	
OVC	6061-T6	40-300	288	30	9.6	0.03	0.04
TS	1100	40-60	63	10	6.3	1.22	15.25
Upper TT	TiAl6V4	4-40	1132	390	2.9	1.63	20.98
Lower TT	TiAl6V4	40-300	1926	338	5.7	0.94	14.01
Coil former	A356-T61	4	330	78	4.2	2.54	23.70
Coils	NbTi composite	4	190	34	5.6	2.51	23.60

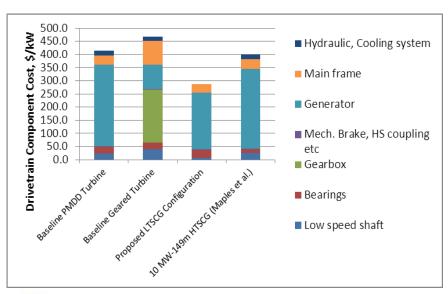


Thermal barrier concept

LTSC generator COE

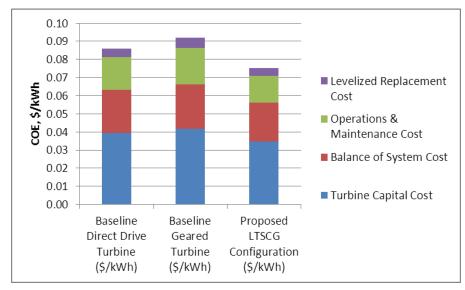
Drivetrain Capex

- LTSC Generator allows...
 - Increasing turbine size to 10MW with reducing drivetrain cost (\$/kW) by 30% over PMDD, 38% over Geared, 28% over HTSCG
 - PMDD cost based on 2010 Maples et al. (NREL/TP-5000-49086), which assumes 2010 rare-earth material prices. Actual PMDD generator costs much higher today.



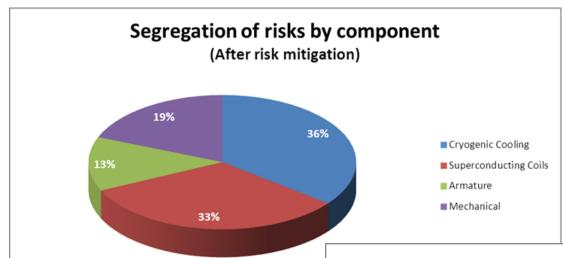
Cost of Energy

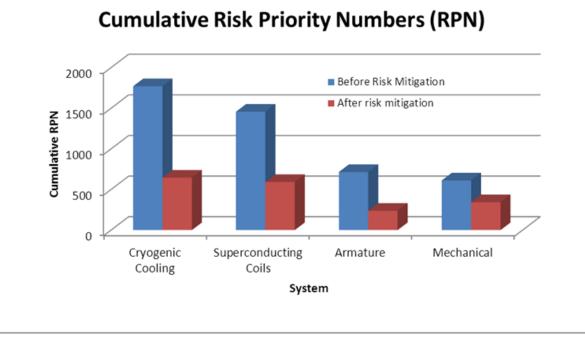
- Baseline is 5MW-126m
- Proposed LTSC Gen is 10MW-160m
- COE Reduction
 - 13% reduction from PMDD, potentially higher due to increased PMDD cost in last 2 years, further potential to reduce SC wire cost
 - 18% reduction over geared





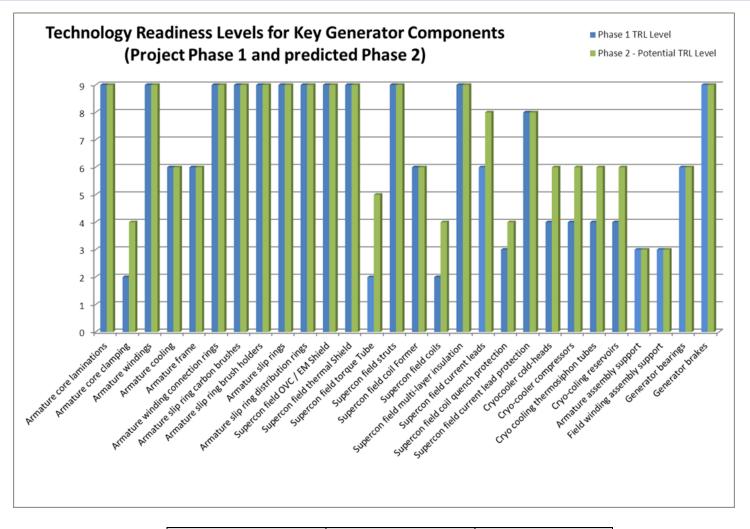
IEEE/CSC & ESAS EUROPEAN SUPERCONDUCTIVITY NEWS FORUM, No. 22, October/November 2012. Component risk identification







IEEE/CSC & ESAS EUROPEAN SUPERCONDUCTIVITY NEWS FORUM, No. 22, October/November 2012. Technology Readiness Level Analysis



(gg)	imagination at work
(36)	imagination at work

	PHASE 1	PHASE 2 projected
Sub-System	% lower than TRL4	% lower than TRL4
Armature	10 %	0 %
Superconducting Field	28 %	0 %
Cryogenic Cooling	0 %	0 %
Mechanical	43%	28 %

Conclusions

- ☐ Superconductivity is competing against well-established and well-understood technology.
- ☐ The political pressures already exist to reduce the cost of energy and to minimize the effect on the environment.
- ☐ Until we can get systems out there working in the 'real world', we will never get sufficient data to be able to prove once and for all that this technology can be the answer to many of our energy-related problems.



Team acknowledgements

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Thank you.



